

Introduction

LEARNING OBJECTIVES

After completing this chapter, you should be able to:

- identify the dimensions of physical quantities and correctly use units systems.
- identify the key fluid properties used in the analysis of fluid behavior.
- calculate values for common fluid properties given appropriate information.
- explain the effects of fluid compressibility.
- use the concepts of viscosity, vapor pressure, and surface tension.

Fluid mechanics is the discipline within the broad field of applied mechanics that is concerned with the behavior of liquids and gases at rest or in motion. It covers a vast array of phenomena that occur in nature (with or without human intervention), in biology, and in numerous engineered, invented, or manufactured situations. There are few aspects of our lives that do not involve fluids, either directly or indirectly.

The immense range of different flow conditions is mind-boggling and strongly dependent on the value of the numerous parameters that describe fluid flow. Among the long list of parameters involved are (1) the physical size of the flow, ℓ ; (2) the speed of the flow, V ; and (3) the pressure, p , as indicated in the adjacent figure for a light aircraft parachute recovery system. These are just three of the important parameters that, along with many others, are discussed in detail in various sections of this book. To get an inkling of the range of some of the parameter values encountered in fluid mechanics, consider the following.

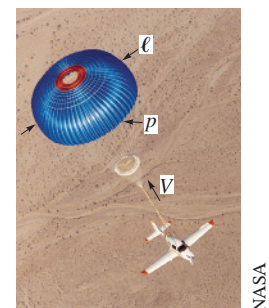
- Size, ℓ

Every flow has a characteristic (or typical) length associated with it. For example, for flow of fluid within pipes, the pipe diameter is a characteristic length. Pipe flows include the flow of water in the pipes in our homes, the blood flow in our arteries and veins, and the airflow in our bronchial tree. They also involve pipe sizes that are not within our everyday experiences. Such examples include the flow of oil across Alaska through a 4-foot-diameter, 799-mile-long pipe and, at the other end of the size scale, the new area of interest involving flow in nanoscale pipes whose diameters are on the order of 10^{-8} m. Each of these pipe flows has important characteristics that are not found in the others.

Characteristic lengths of some other flows are shown in **Fig. 1.1a**.

- Speed, V

As we note from The Weather Channel, on a given day the wind speed may cover what we think of as a wide range, from a gentle 5-mph breeze to a 100-mph hurricane or a 250-mph tornado. However, this speed range is small compared to that of the almost





V1.1 Mt. St. Helens eruption

imperceptible flow of the fluid-like magma below the Earth's surface that drives the continental drift motion of the tectonic plates at a speed of about 2×10^{-8} m/s or the hypersonic airflow around a meteor as it streaks through the atmosphere at 3×10^4 m/s.

Characteristic speeds of some other flows are shown in **Fig. 1.1b**.

- Pressure, p

The pressure within fluids covers an extremely wide range of values. We are accustomed to the 35 psi (lb/in.²) pressure within our car's tires, the "120 over 70" typical blood pressure reading, or the standard 14.7 psi atmospheric pressure. However, the large 10,000 psi pressure in the hydraulic ram of an earth mover or the tiny 2×10^{-6} psi pressure of a sound wave generated at ordinary talking levels are not easy to comprehend.

Characteristic pressures of some other flows are shown in **Fig. 1.1c**.

The list of fluid mechanics applications goes on and on. But you get the point. Fluid mechanics is a very important, practical subject that encompasses a wide variety of situations. It is very likely that during your career as an engineer you will be involved in the analysis and design of systems that require a good understanding of fluid mechanics. Although it is not possible to adequately cover all of the important areas of fluid mechanics within one book, it is hoped that this introductory text will provide a sound foundation of the fundamental aspects of fluid mechanics.



V1.2 E. coli swimming

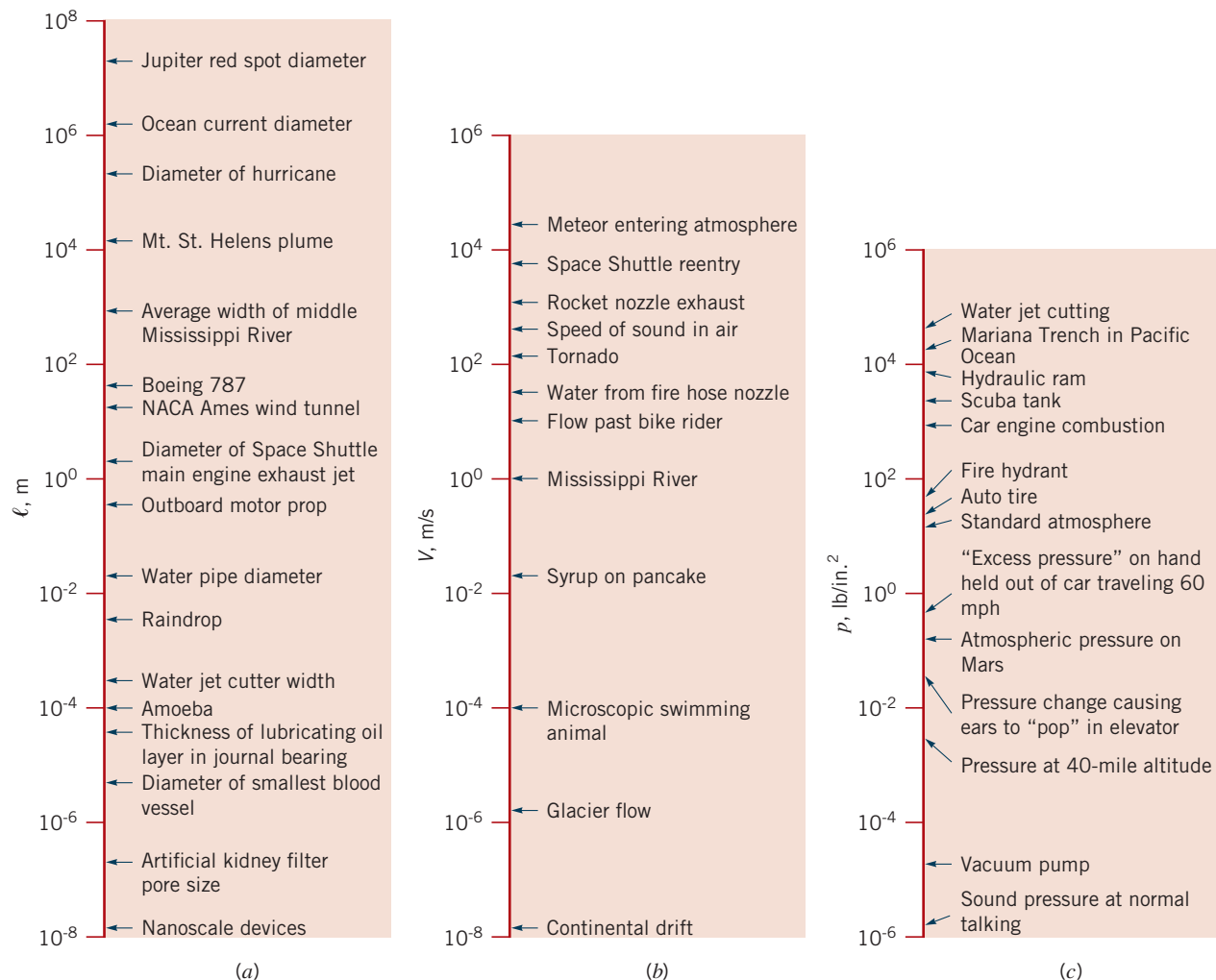


FIGURE 1.1 Characteristic values of some fluid flow parameters for a variety of flows: (a) object size, (b) fluid speed, (c) fluid pressure.

1.1 Some Characteristics of Fluids

One of the first questions we need to explore is—what is a fluid? Or we might ask—what is the difference between a solid and a fluid? We have a general, vague idea of the difference. A solid is “hard” and not easily deformed, whereas a fluid is “soft” and is easily deformed (we can readily move through air). Although quite descriptive, these casual observations of the differences between solids and fluids are not very satisfactory from a scientific or engineering point of view. A closer look at the molecular structure of materials reveals that matter that we commonly think of as a solid (steel, concrete, etc.) has densely spaced molecules with large intermolecular cohesive forces that allow the solid to maintain its shape, and to not be easily deformed. However, for matter that we normally think of as a liquid (water, oil, etc.), the molecules are spaced farther apart, the intermolecular forces are smaller than for solids, and the molecules have more freedom of movement. Thus, liquids can be easily deformed (but not easily compressed) and can be poured into containers or forced through a tube. Gases (air, oxygen, etc.) have even greater molecular spacing and freedom of motion with negligible cohesive intermolecular forces and, as a consequence, are easily deformed (and compressed) and will completely fill the volume of any container in which they are placed. If you drop a block of steel into a container, its shape does not change. If you pour water into a container, it assumes the shape of the container, but its volume does not change. If you introduce air into a container, it expands to fill the container. Liquids and gases are both fluids.

Both liquids and gases are fluids.

The Wide World of Fluids

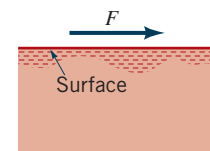
Will what works in air work in water?

For the past few years, a San Francisco company has been working on small, maneuverable submarines designed to travel through water using wings, controls, and thrusters that are similar to those on jet airplanes. After all, water (for submarines) and air (for airplanes) are both fluids, so it is expected that many of the principles governing the flight of airplanes should carry over to the “flight” of winged submarines. Of course, there are

differences. For example, the submarine must be designed to withstand external pressures of nearly 700 pounds per square inch greater than that inside the vehicle. On the other hand, at high altitude where commercial jets fly, the exterior pressure is 3.5 psi rather than standard sea-level pressure of 14.7 psi, so the vehicle must be pressurized internally for passenger comfort. In both cases, however, the design of the craft for minimal drag, maximum lift, and efficient thrust is governed by the same fluid dynamic concepts.

Although the differences between solids and fluids can be explained qualitatively on the basis of molecular structure, a more specific distinction is based on how they deform under the action of an external load. Specifically, a **fluid** is defined as a substance that deforms continuously when acted on by a shearing stress of any magnitude. A shearing stress (force per unit area) is created whenever a tangential force acts on a surface, as shown in the adjacent figure. When common solids such as steel or other metals are acted on by a shearing stress, they will initially deform (usually a very small deformation), but they will not continuously deform (flow). However, common fluids such as water, oil, and air satisfy the definition of a fluid—that is, they will flow when acted on by a shearing stress. Some materials, such as slurries, tar, putty, toothpaste, and so on, are not easily classified since they will behave as a solid if the applied shearing stress is small, but if the stress exceeds some critical value, the substance will flow. The study of such materials is called *rheology* and does not fall within the province of classical fluid mechanics. Thus, all the fluids we will be concerned with in this text will conform to the definition of a fluid.

Although the molecular structure of fluids is important in distinguishing one fluid from another, it is not yet practical to study the behavior of individual molecules when trying to describe the behavior of fluids at rest or in motion. Rather, we characterize the behavior by considering the average, or macroscopic, value of the quantity of interest, where the average is evaluated over a small volume containing a large number of molecules. Thus, when we say that the velocity at a certain point in a fluid is so much, we are really indicating the average



velocity of the molecules in a small volume surrounding the point. The volume is small compared with the physical dimensions of the system of interest, but large compared with the average distance between molecules. Is this a reasonable way to describe the behavior of a fluid? The answer is generally yes, since the spacing between molecules is typically very small. For gases at normal pressures and temperatures, the spacing is on the order of 10^{-6} mm, and for liquids it is on the order of 10^{-7} mm. The number of molecules per cubic millimeter is on the order of 10^{18} for gases and 10^{21} for liquids. It is clear that the number of molecules in a very tiny volume is huge, and the idea of using average values taken over this volume is certainly reasonable. We assume that all the fluid characteristics we are interested in (pressure, velocity, etc.) vary continuously throughout the fluid—that is, we treat the fluid as a *continuum* and we refer to the very small volume as a point in the flow. This concept will certainly be valid for all the circumstances considered in this text. One area of fluid mechanics for which the continuum concept breaks down is in the study of rarefied gases such as would be encountered at very high altitudes. In this case, the spacing between air molecules can become large and continuum mechanics is not a good model of reality for these flows.

1.2 Dimensions, Dimensional Homogeneity, and Units

Since in our study of fluid mechanics we will be dealing with a variety of fluid characteristics, it is necessary to develop a system for describing these characteristics both *qualitatively* and *quantitatively*. The qualitative aspect serves to identify the nature, or type, of the characteristics (such as length, time, stress, and velocity), whereas the quantitative aspect provides a numerical measure of the characteristics. The quantitative description requires both a number and a standard by which various quantities can be compared. A standard for length might be a meter or foot, for time an hour or second, and for mass a slug or kilogram. Such standards are called **units**, and several systems of units are in common use as described in the following section. The qualitative description is conveniently given in terms of *primary quantities*, such as length, L , time, T , mass, M , and temperature, Θ . These primary quantities can then be used to provide a qualitative description of any other *secondary quantity*: for example, area $\doteq L^2$, velocity $\doteq LT^{-1}$, density $\doteq ML^{-3}$, and so on, where the symbol $\doteq$ is used to indicate the *dimensions* of the secondary quantity in terms of the primary quantities. Thus, to describe qualitatively a velocity, V , we would write

$$V \doteq LT^{-1}$$

and say that “the dimensions of a velocity equal length divided by time.” The primary quantities are also referred to as **basic dimensions**.

For a wide variety of problems involving fluid mechanics, only the three basic dimensions, L , T , and M are required. Alternatively, L , T , and F could be used, where F is the basic dimensions of force. Since Newton’s law states that force is equal to mass times acceleration, it follows that $F \doteq MLT^{-2}$ or $M \doteq FL^{-1}T^2$. Thus, secondary quantities expressed in terms of M can be expressed in terms of F through the relationship above. For example, stress, σ , is a force per unit area, so that $\sigma \doteq FL^{-2}$, but an equivalent dimensional equation is $\sigma \doteq ML^{-1}T^{-2}$. **Table 1.1** provides a list of dimensions for a number of common physical quantities.

All theoretically derived equations are **dimensionally homogeneous**—that is, the dimensions of the left side of the equation must be the same as those on the right side, and all additive terms must have the same dimensions. We accept as a fundamental premise that all equations describing physical phenomena must be dimensionally homogeneous. If this were not true, we would be attempting to equate or add unlike physical quantities, which would not make sense. For example, the equation for the velocity, V , of a uniformly accelerated body is

$$V = V_0 + at \tag{1.1}$$

Fluid characteristics can be described qualitatively in terms of certain basic quantities such as length, time, and mass.

TABLE 1.1 Dimensions Associated with Common Physical Quantities

	<i>FLT</i> System	<i>MLT</i> System			<i>FLT</i> System	<i>MLT</i> System
Acceleration	LT^{-2}	LT^{-2}		Power	FLT^{-1}	ML^2T^{-3}
Angle	$F^0L^0T^0$	$M^0L^0T^0$		Pressure	FL^{-2}	$ML^{-1}T^{-2}$
Angular acceleration	T^{-2}	T^{-2}		Specific heat	$L^2T^{-2}\Theta^{-1}$	$L^2T^{-2}\Theta^{-1}$
Angular velocity	T^{-1}	T^{-1}		Specific weight	FL^{-3}	$ML^{-2}T^{-2}$
Area	L^2	L^2		Strain	$F^0L^0T^0$	$M^0L^0T^0$
Density	$FL^{-4}T^2$	ML^{-3}		Stress	FL^{-2}	$ML^{-1}T^{-2}$
Energy	FL	ML^2T^{-2}		Surface tension	FL^{-1}	MT^{-2}
Force	F	MLT^{-2}		Temperature	Θ	Θ
Frequency	T^{-1}	T^{-1}		Time	T	T
Heat	FL	ML^2T^{-2}		Torque	FL	ML^2T^{-2}
Length	L	L		Velocity	LT^{-1}	LT^{-1}
Mass	$FL^{-1}T^2$	M		Viscosity (dynamic)	$FL^{-2}T$	$ML^{-1}T^{-1}$
Modulus of elasticity	FL^{-2}	$ML^{-1}T^{-2}$		Viscosity (kinematic)	L^2T^{-1}	L^2T^{-1}
Moment of a force	FL	ML^2T^{-2}		Volume	L^3	L^3
Moment of inertia (area)	L^4	L^4		Work	FL	ML^2T^{-2}
Moment of inertia (mass)	FLT^2	ML^2				
Momentum	FT	MLT^{-1}				

where V_0 is the initial velocity, a the acceleration, and t the time interval. In terms of dimensions the equation is

$$LT^{-1} \doteq LT^{-1} + LT^{-2}T$$

and thus Eq. 1.1 is dimensionally homogeneous.

Some equations that are known to be valid contain constants having dimensions. The equation for the distance, d , traveled by a freely falling body can be written as

$$d = 16.1t^2 \quad (1.2)$$

and a check of the dimensions reveals that the constant must have the dimensions of LT^{-2} if the equation is to be dimensionally homogeneous. Actually, Eq. 1.2 is a special form of the well-known equation from physics for freely falling bodies,

$$d = \frac{gt^2}{2} \quad (1.3)$$

in which g is the acceleration of gravity. Equation 1.3 is dimensionally homogeneous and valid in any system of units. For $g = 32.2 \text{ ft/s}^2$ the equation reduces to Eq. 1.2, and thus Eq. 1.2 is valid only for the system of units using feet and seconds. Equations that are restricted to a particular system of units can be denoted as *restricted homogeneous equations*, as opposed to equations valid in any system of units, which are *general homogeneous equations*. The preceding discussion indicates one rather elementary, but important, use of the concept of dimensions: the determination of one aspect of the generality of a given equation simply based on a consideration of the dimensions of the various terms in the equation. The concept of dimensions also forms the basis for the powerful tool of *dimensional analysis*, which is considered in detail in Chapter 7.

General homogeneous equations are valid in any system of units.

Note to the Users of This Text All of the examples in the text use a consistent problem-solving methodology, which is similar to that in other engineering courses such as statics. Each example highlights the key elements of analysis: **Given**, **Find**, **Solution**, and **Comment**.

The **Given** and **Find** steps clearly identify the information specified to define the problem.

The **Solution** step is where the equations needed to solve the problem are formulated and the problem is actually solved. In this step, there are typically several other tasks that help to set up the solution and are required to solve the problem. The first is a drawing of the problem; where appropriate, it is always helpful to draw a sketch of the problem. Here, the relevant geometry and coordinate system to be used as well as features such as control volumes, forces and pressures, velocities, and mass flowrates are included. This helps in gaining an understanding of the problem. Making appropriate assumptions to solve the problem is the second task. In a realistic engineering problem-solving environment, the appropriate assumptions are developed as an integral part of the solution process. Assumptions can provide simplifications or offer useful constraints, both of which can help in solving the problem. Throughout the examples in this text, the appropriate assumptions are embedded within the **Solution** step, as they are in solving a real-world problem. This provides a realistic problem-solving experience.

The final element in the methodology is the **Comment**. For the examples in the text, this section is used to provide further insight into the problem or the solution. It can also be a point in the analysis at which certain questions are posed. For example: Is the answer reasonable, and does it make physical sense? Are the final units correct? If the value of a parameter was changed, how would the answer change? Adopting this methodology will aid in the development of problem-solving skills for fluid mechanics, as well as other engineering disciplines.

EXAMPLE 1.1 | Restricted and General Homogeneous Equations

Given A liquid flows through an orifice located in the side of a tank, as shown in **Fig. E1.1**. A commonly used equation for determining the volume rate of flow, Q , through the orifice is

$$Q = 0.61 A \sqrt{2gh}$$

where A is the area of the orifice, g is the acceleration of gravity, and h is the height of the liquid above the orifice.

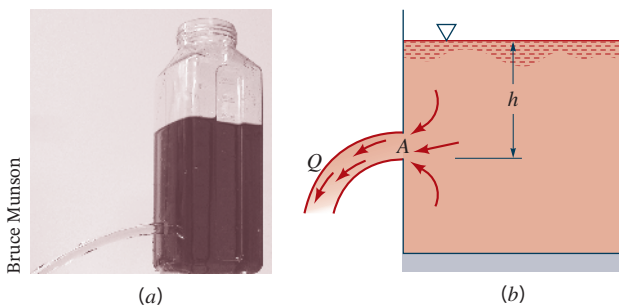


FIGURE E1.1

Find Investigate the dimensional homogeneity of this formula.

Solution The dimensions of the various terms in the equation are $Q = \text{volume/time} \doteq L^3 T^{-1}$, $A = \text{area} \doteq L^2$, $g = \text{acceleration of gravity} \doteq L T^{-2}$, and $h = \text{height} \doteq L$.

These terms, when substituted into the equation, yield the dimensional form:

$$(L^3 T^{-1}) \doteq (0.61)(L^2)(\sqrt{2})(L T^{-2})^{1/2}(L)^{1/2}$$

or

$$(L^3 T^{-1}) \doteq [0.61 \sqrt{2}](L^3 T^{-1})$$

It is clear from this result that the equation is dimensionally homogeneous (both sides of the formula have the same dimensions of $L^3 T^{-1}$), and the number $0.61 \sqrt{2}$ is dimensionless.

If we were going to use this relationship repeatedly, we might be tempted to simplify it by replacing g with its standard value of 32.2 ft/s^2 and rewriting the formula as

$$Q = 4.90 A \sqrt{h} \quad (1)$$

A quick check of the dimensions reveals that

$$L^3 T^{-1} \doteq (4.90)(L^{5/2})$$

and, therefore, the equation expressed as Eq. 1 can only be dimensionally correct if the number 4.90 has the dimensions of $L^{1/2} T^{-1}$. Whenever a number appearing in an equation or formula has dimensions, it means that the specific value of the number will depend on the system of units used. Thus, for the case being considered with feet and seconds used as units, the number 4.90 has units of $\text{ft}^{1/2}/\text{s}$. Equation 1 will only give the correct value for Q (in ft^3/s) when A is expressed in square feet and h in feet. Thus, Eq. 1 is a *restricted* homogeneous equation, whereas the original equation is a *general* homogeneous equation that would be valid for any consistent system of units.

Comment A quick check of the dimensions of the various terms in an equation is a useful practice and will often be helpful in detecting errors—that is, as noted previously, all physically meaningful equations must be dimensionally homogeneous. We have briefly alluded to units in this example, and this important topic will be considered in more detail in the next section.

1.2.1 Systems of Units

In addition to the qualitative description of the various quantities of interest, it is generally necessary to have a quantitative measure of a quantity. For example, if we measure the width of this page in the book and say that it is 10 units wide, the statement has no meaning until the unit of length is defined. If we indicate that the unit of length is a meter, and define the meter as some standard length, a unit system for length has been established (and a numerical value can be given to the page width). In addition to length, a unit must be established for each of the remaining basic quantities (force, mass, time, and temperature). There are several systems of units in use, and we shall consider three systems that are commonly used in engineering.

International System (SI, *Système Internationale d'Unités*) In 1960, the Eleventh General Conference on Weights and Measures, the international organization responsible for maintaining precise uniform standards of measurements, formally adopted the *International System of Units* as the international standard. This system, commonly termed SI, has been widely adopted worldwide and is widely used (although certainly not exclusively) in the United States. It is expected that the long-term trend will be for all countries to accept SI as the accepted standard, and it is imperative that engineering students become familiar with this system. In SI the unit of length is the meter (m), the time unit is the second (s), the mass unit is the kilogram (kg), and the temperature unit is the kelvin (K). Note that there is no degree symbol used when expressing a temperature in kelvin units. The kelvin temperature scale is an absolute scale and is related to the Celsius (centigrade) scale ($^{\circ}\text{C}$) through the relationship

$$\text{K} = ^{\circ}\text{C} + 273.15$$

Although the Celsius scale is not in itself part of SI, it is common practice to specify temperatures in degrees Celsius when using SI units.

The force unit, called the newton (N), is defined from Newton's second law as

$$1 \text{ N} = (1 \text{ kg})(1 \text{ m/s}^2)$$

Thus, a 1-N force acting on a 1-kg mass will give the mass an acceleration of 1 m/s^2 . Standard gravity in SI is 9.807 m/s^2 (commonly approximated as 9.81 m/s^2) so that a 1-kg mass weighs 9.81 N under standard gravity. Note that weight and mass are different, both qualitatively and quantitatively! The unit of *work* in SI is the joule (J), which is the work done when the point of application of a 1-N force is displaced through a 1-m distance in the direction of a force. Thus,

$$1 \text{ J} = 1 \text{ N} \cdot \text{m}$$

Power is the rate of doing work. The unit of *power* is the watt (W) defined as a joule per second. Thus,

$$1 \text{ W} = 1 \text{ J/s} = 1 \text{ N} \cdot \text{m/s}$$

Prefixes for forming multiples and fractions of SI units are given in **Table 1.2**. For example, the notation kN is read as “kilonewtons” and stands for 10^3 N . Similarly, mm is read as “millimeters” and stands for 10^{-3} m . The centimeter is not an accepted unit of length in the SI system, so for most problems in fluid mechanics in which SI units are used, lengths will be expressed in millimeters or meters.

British Gravitational (BG) System In the BG system, the unit of length is the foot (ft), the time unit is the second (s), the force unit is the pound (lb), and the temperature unit is the degree Fahrenheit ($^{\circ}\text{F}$) or the absolute temperature unit is the degree Rankine ($^{\circ}\text{R}$), where

$$^{\circ}\text{R} = ^{\circ}\text{F} + 459.67$$

The mass unit, called the *slug*, is defined from Newton's second law (force = mass $\times$ acceleration) as

$$1 \text{ lb} = (1 \text{ slug})(1 \text{ ft/s}^2)$$

In mechanics it is very important to distinguish between weight and mass.

TABLE 1.2 Prefixes for SI Units

Factor by Which Unit Is Multiplied	Prefix	Symbol	Factor by Which Unit Is Multiplied	Prefix	Symbol
10^{15}	peta	P	10^{-2}	centi	c
10^{12}	tera	T	10^{-3}	milli	m
10^9	giga	G	10^{-6}	micro	μ
10^6	mega	M	10^{-9}	nano	n
10^3	kilo	k	10^{-12}	pico	p
10^2	hecto	h	10^{-15}	femto	f
10	deka	da	10^{-18}	atto	a
10^{-1}	deci	d			

This relationship indicates that a 1-lb force acting on a mass of 1 slug will give the mass an acceleration of 1 ft/s^2 .

The weight, W (which is the force due to gravity, g), of a mass, m , is given by the equation

$$W = mg$$

and in BG units

$$W (\text{lb}) = m (\text{slugs}) g (\text{ft/s}^2)$$

Since Earth's standard gravity is taken as $g = 32.174 \text{ ft/s}^2$ (commonly approximated as 32.2 ft/s^2), it follows that a mass of 1 slug weighs 32.2 lb under standard gravity.

Three units systems are widely used in engineering:
BG – British Gravitational
EE – English Engineering
SI – International System

The Wide World of Fluids

How long is a foot?

Today, in the United States, the common length *unit* is the *foot*, but throughout antiquity the unit used to measure length has quite a history. The first length units were based on the lengths of various body parts. One of the earliest units was the Egyptian cubit, first used around 3000 B.C. and defined as the length of the arm from elbow to extended fingertips. Other measures followed, with the foot simply taken as the length of a man's foot. Since this length obviously varies from person to person it was often "standardized"

by using the length of the current reigning royalty's foot. In 1791, a special French commission proposed that a new universal length unit called a meter (metre) be defined as the distance of one-quarter of the Earth's meridian (north pole to the equator) divided by 10 million. Although controversial, the meter was accepted in 1799 as the standard. With the development of advanced technology, the length of a meter was redefined in 1983 as the distance traveled by light in a vacuum during the time interval of $1/299,792,458 \text{ s}$. The foot is now defined as 0.3048 meter. Our simple rulers and yardsticks indeed have an intriguing history.

English Engineering (EE) System In the EE system, units for force *and* mass are defined independently; thus, special care must be exercised when using this system in conjunction with Newton's second law. The basic unit of mass is the pound mass (lbm), and the unit of force is the pound (lb).¹ The unit of length is the foot (ft), the unit of time is the second (s), and the absolute temperature scale is the degree Rankine ($^{\circ}\text{R}$). To make the equation expressing Newton's second law dimensionally homogeneous, we write it as

$$\mathbf{F} = \frac{m\mathbf{a}}{g_c} \quad (1.4)$$

where g_c is a constant of proportionality, which allows us to define units for both force and mass. For the BG system, only the force unit was prescribed and the mass unit defined in a consistent manner such that $g_c = 1$. Similarly, for SI the mass unit was prescribed and the force unit defined in a consistent manner such that $g_c = 1$. For the EE system, a 1-lb force is defined as the force which accelerates a 1 lbm at a rate equal to the standard acceleration due to the earth's gravity, $\approx 32.2 \text{ ft/s}^2$.

¹It is also common practice to use the notation, lbf, to indicate pound force.

Thus, for Eq. 1.4 to be both numerically and dimensionally correct

$$1 \text{ lb} = \frac{(1 \text{ lbm})(32.174 \text{ ft/s}^2)}{g_c}$$

so that

$$g_c = \frac{(1 \text{ lbm})(32.174 \text{ ft/s}^2)}{(1 \text{ lb})}$$

With the EE system, weight and mass are related through the equation

$$W = \frac{mg}{g_c}$$

where g is the local acceleration of gravity. Under conditions of standard gravity ($g = g_c$), the weight in pounds and the mass in pound mass are numerically equal. Also, since a 1-lb force gives a mass of 1 lbm an acceleration of 32.174 ft/s^2 and a mass of 1 slug an acceleration of 1 ft/s^2 , it follows that

$$1 \text{ slug} = 32.174 \text{ lbm}$$

We cannot overemphasize the importance of paying close attention to units when solving problems. It is very easy to introduce huge errors into problem solutions through the use of incorrect units. Get in the habit of using a *consistent* system of units throughout a given solution. It really makes no difference which system you use as long as you are consistent; for example, don't mix slugs and newtons. If problem data are specified in SI units, then use SI units throughout the solution. If the data are specified in BG units, then use BG units throughout the solution. The relative sizes of the SI, BG, and EE units of length, mass, and force are shown in **Fig. 1.2**.

Extensive tables of conversion factors between unit systems, and within unit systems, are provided in Appendix E. For your convenience, abbreviated tables of conversion factors for some quantities commonly encountered in fluid mechanics are presented in **Tables 1.3** and **1.4**, which immediately follow the front cover of this text (using a slightly different format than Appendix E). Note that numbers in these tables are presented in computer exponential notation. For example, the number $5.154 \text{ E}+2$ is the number 5.154×10^2 in scientific notation. You should note that each conversion factor can be thought of as a fraction in which the numerator and denominator are equivalent. For example, an entry for "Length" from Table 1.4 instructs the user "To convert from ... m ... to ... ft ... Multiply by 3.281." Therefore 1 m is the same length as 3.281 ft. Therefore, a fraction formed with a numerator of 1 m and a denominator of 3.281 ft is the very definition of a fraction with a value of one, as is its reciprocal. This may seem obvious when the units of the denominator and numerator are of the same dimension. It is equally true for the more complicated conversion factors that include multiple dimensions and therefore a greater number of units. You already know that you can multiply any quantity by one without changing its value. Likewise, you can multiply (or divide) any quantity by any conversion factor in the tables, provided you use both the number and the units. The result will be correct, even if it does not yield the result you hoped for.

When solving problems it is important to use a consistent system of units, e.g., don't mix BG and SI units.

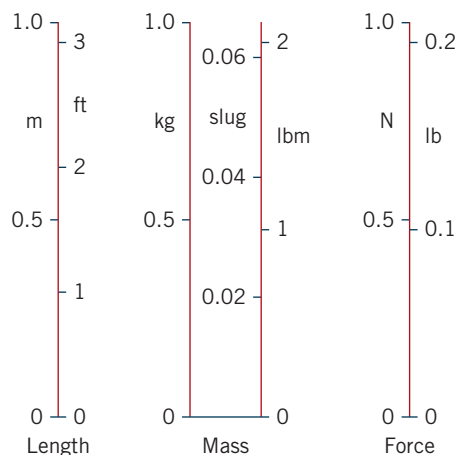


FIGURE 1.2 Comparison of SI, BG, and EE units.

TABLE 1.3 Conversion Factors from BG and EE Units to SI Units^a

	To Convert from	to	Multiply by
Acceleration	ft/s ²	m/s ²	3.048 E − 1
Area	ft ²	m ²	9.290 E − 2
Density	lbm/ft ³	kg/m ³	1.602 E + 1
	slugs/ft ³	kg/m ³	5.154 E + 2
Energy	Btu	J	1.055 E + 3
	ft · lb	J	1.356
Force	lb	N	4.448
Length	ft	m	3.048 E − 1
	in.	m	2.540 E − 2
	mile	m	1.609 E + 3
Mass	lbm	kg	4.536 E − 1
	slug	kg	1.459 E + 1
Power	ft · lb/s	W	1.356
	hp	W	7.457 E + 2
Pressure	in. Hg (60 °F)	N/m ²	3.377 E + 3
	lb/ft ² (psf)	N/m ²	4.788 E + 1
	lb/in. ² (psi)	N/m ²	6.895 E + 3
Specific weight	lb/ft ³	N/m ³	1.571 E + 2
Temperature	°F	°C	$T_C = (5/9)(T_F - 32^\circ)$
	°R	K	5.556 E − 1
Velocity	ft/s	m/s	3.048 E − 1
	mi/hr (mph)	m/s	4.470 E − 1
Viscosity (dynamic)	lb · s/ft ²	N · s/m ²	4.788 E + 1
Viscosity (kinematic)	ft ² /s	m ² /s	9.290 E − 2
Volume flowrate	ft ³ /s	m ³ /s	2.832 E − 2
	gal/min (gpm)	m ³ /s	6.309 E − 5

^aFor more unit conversions, refer to Appendix E.**TABLE 1.4** Conversion Factors from SI Units to BG and EE Units^a

	To Convert from	to	Multiply by
Acceleration	m/s ²	ft/s ²	3.281
Area	m ²	ft ²	1.076 E + 1
Density	kg/m ³	lbm/ft ³	6.243 E − 2
	kg/m ³	slugs/ft ³	1.940 E − 3
Energy	J	Btu	9.478 E − 4
	J	ft · lb	7.376 E − 1
Force	N	lb	2.248 E − 1
Length	m	ft	3.281
	m	in.	3.937 E + 1
	m	mile	6.214 E − 4
Mass	kg	lbm	2.205
	kg	slug	6.852 E − 2
Power	W	ft · lb/s	7.376 E − 1
	W	hp	1.341 E − 3
Pressure	N/m ²	in. Hg (60 °F)	2.961 E − 4
	N/m ²	lb/ft ² (psf)	2.089 E − 2
	N/m ²	lb/in. ² (psi)	1.450 E − 4
Specific weight	N/m ³	lb/ft ³	6.366 E − 3
Temperature	°C	°F	$T_F = 1.8 T_C + 32^\circ$
	K	°R	1.800
Velocity	m/s	ft/s	3.281
	m/s	mi/hr (mph)	2.237

(continue)

TABLE 1.4 Conversion Factors from SI Units to BG and EE Units^a (continued)

	To Convert from	to	Multiply by
Viscosity (dynamic)	$\text{N} \cdot \text{s}/\text{m}^2$	$\text{lb} \cdot \text{s}/\text{ft}^2$	$2.089 \text{ E} - 2$
Viscosity (kinematic)	m^2/s	ft^2/s	$1.076 \text{ E} + 1$
Volume flowrate	m^3/s	ft^3/s	$3.531 \text{ E} + 1$
	m^3/s	$\text{gal}/\text{min} \text{ (gpm)}$	$1.585 \text{ E} + 4$

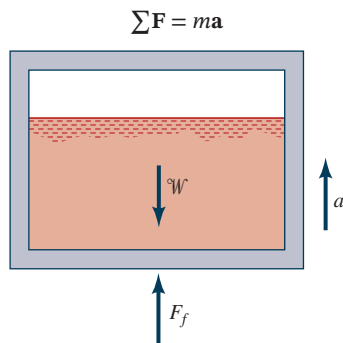
^aFor more unit conversions, refer to Appendix E.**EXAMPLE 1.2** | BG and SI Units

Given A tank of liquid having a total mass of 36 kg rests on a support in the equipment bay of the Falcon Heavy.

Find Determine the force (in newtons) that the tank exerts on the support shortly after lift off when the rocket is accelerating upward as shown in **Fig. E1.2a** at $15 \text{ ft}/\text{s}^2$.

**FIGURE E1.2a**

Solution A free-body diagram of the tank is shown in **Fig. E1.2b**, where $\mathcal{W}$ is the weight of the tank and liquid, and F_f is the reaction of the floor on the tank. Application of Newton's second law of motion to this body gives

**FIGURE E1.2b**

or

$$F_f - \mathcal{W} = ma \quad (1)$$

where we have taken upward as the positive direction. Since $\mathcal{W} = mg$, Eq. 1 can be written as

$$F_f = m(g + a) \quad (2)$$

Before substituting any number into Eq. 2, we must decide on a system of units and then be sure all of the data are expressed in these units. Since we want F_f in newtons, we will use SI units so that

$$\begin{aligned} F_f &= 36 \text{ kg} [9.81 \text{ m}/\text{s}^2 + (15 \text{ ft}/\text{s}^2)(0.3048 \text{ m}/\text{ft})] \\ &= 518 \text{ kg} \cdot \text{m}/\text{s}^2 \end{aligned}$$

Since $1 \text{ N} = 1 \text{ kg} \cdot \text{m}/\text{s}^2$, it follows that

$$F_f = 518 \text{ N} \quad (\text{downward on floor}) \quad (\text{Ans})$$

The direction is downward, since the force shown on the free-body diagram is the force of the support *on the tank* so that the force the tank exerts *on the support* is equal in magnitude but opposite in direction.

Comment As you work through a large variety of problems in this text, you will find that units play an essential role in arriving at a numerical answer. Be careful! It is easy to mix units and cause large errors. If in the above example the acceleration had been left as $15 \text{ ft}/\text{s}^2$ with m and g expressed in SI units, we would have calculated the force as 893 N and the answer would have been 72% too large!

The Wide World of Fluids**Units and space travel**

A NASA spacecraft, the Mars Climate Orbiter, was launched in December 1998 to study the Martian geography and weather patterns. The spacecraft was slated to begin orbiting Mars on September 23, 1999. However, NASA officials lost communication with the spacecraft early that day and it is believed that the spacecraft broke

apart or overheated because it came too close to the surface of Mars. Errors in the maneuvering commands sent from Earth caused the Orbiter to sweep within 37 miles of the surface rather than the intended 93 miles. The subsequent investigation revealed that the errors were due to a simple mix-up in *units*. One team controlling the Orbiter used SI units, whereas another team used BG units. This costly experience illustrates the importance of using a consistent system of units.

1.3 Analysis of Fluid Behavior

The study of fluid mechanics involves the same fundamental laws and principles you have encountered in physics and other mechanics courses. These laws include Newton's laws of motion, conservation of mass, and the first and second laws of thermodynamics. Thus, there are strong similarities between the general approach to fluid mechanics and to rigid-body and deformable-body solid mechanics. This is indeed helpful since many of the concepts and techniques of analysis used in fluid mechanics will be ones you have encountered before in other courses.

The broad subject of fluid mechanics can be generally subdivided into *fluid statics*, in which the fluid is at rest, and *fluid dynamics*, in which the fluid is moving. In the following chapters, we will consider both of these areas in detail. Before we can proceed, however, it will be necessary to define and discuss certain fluid *properties* that are intimately related to fluid behavior. It is obvious that different fluids can have grossly different characteristics. For example, gases are light and compressible, whereas liquids are heavy (by comparison) and relatively incompressible. A syrup flows slowly from a container, but water flows rapidly when poured from the same container. To quantify these differences, certain fluid properties are used. In the following several sections, properties that play an important role in the analysis of fluid behavior are considered.

1.4 Measures of Fluid Mass and Weight

1.4.1 Density

The density of a fluid is defined as its mass per unit volume.

The **density** of a fluid, designated by the Greek symbol ρ (rho), is defined as its mass per unit volume. Density is typically used to characterize the mass of a fluid system. In the BG system, ρ has units of slugs/ft³, in EE the units are lb_m/ft³, and in SI the units are kg/m³.

The value of density can vary widely between different fluids, but for liquids, variations in pressure and temperature generally have only a small effect on the value of ρ . The small change in the density of water with large variations in temperature is illustrated in **Fig. 1.3**. **Tables 1.5** and **1.6** list values of density for several common liquids. The density of water at 60 °F is 1.94 slugs/ft³ or 999 kg/m³. The large numerical difference between those two values illustrates the importance of paying attention to units! Unlike liquids, the density of a gas is strongly influenced by both pressure and temperature, and this difference will be discussed in the next section.

The *specific volume*, v , is the *volume* per unit mass and is therefore the reciprocal of the density—that is,

$$v = \frac{1}{\rho} \quad (1.5)$$

This property is commonly used in thermodynamics but not in fluid mechanics.

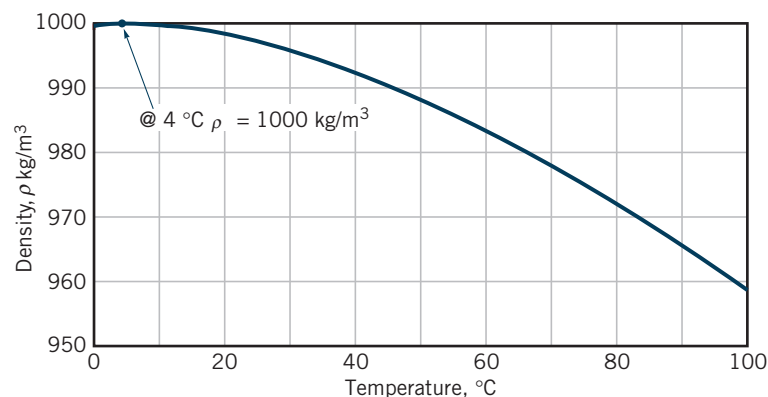


FIGURE 1.3 Density of water as a function of temperature.

TABLE 1.5 Approximate Physical Properties of Some Common Liquids (BG Units)

Liquid	Temperature (°F)	Density, ρ (slugs/ft ³)	Specific Weight, γ (lb/ft ³)	Dynamic Viscosity, μ (lb · s/ft ²)	Kinematic Viscosity, ν (ft ² /s)	Surface Tension, ^a σ (lb/ft)	Vapor Pressure, p_v [lb/in. ² (abs)]	Bulk Modulus, ^b E_v (lb/in. ²)
Carbon tetrachloride	68	3.09	99.5	2.00 E − 5	6.47 E − 6	1.84 E − 3	1.9 E + 0	1.91 E + 5
Ethyl alcohol	68	1.53	49.3	2.49 E − 5	1.63 E − 5	1.56 E − 3	8.5 E − 1	1.54 E + 5
Gasoline ^c	60	1.32	42.5	6.5 E − 6	4.9 E − 6	1.5 E − 3	8.0 E + 0	1.9 E + 5
Glycerin	68	2.44	78.6	3.13 E − 2	1.28 E − 2	4.34 E − 3	2.0 E − 6	6.56 E + 5
Mercury	68	26.3	847	3.28 E − 5	1.25 E − 6	3.19 E − 2	2.3 E − 5	4.14 E + 6
SAE 30 oil ^c	60	1.77	57.0	8.0 E − 3	4.5 E − 3	2.5 E − 3	—	2.2 E + 5
Seawater	60	1.99	64.0	2.51 E − 5	1.26 E − 5	5.03 E − 3	2.56 E − 1	3.39 E + 5
Water	60	1.94	62.4	2.34 E − 5	1.21 E − 5	5.03 E − 3	2.56 E − 1	3.12 E + 5

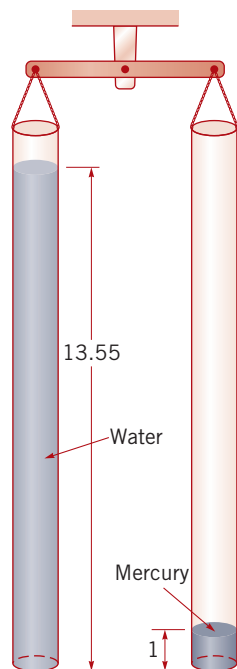
^aIn contact with air.
^bIsentropic bulk modulus calculated from speed of sound.
^cTypical values. Properties of petroleum products vary.

TABLE 1.6 Approximate Physical Properties of Some Common Liquids (SI Units)

Liquid	Temperature (°C)	Density, ρ (kg/m ³)	Specific Weight, γ (kN/m ³)	Dynamic Viscosity, μ (N · s/m ²)	Kinematic Viscosity, ν (m ² /s)	Surface Tension, ^a σ (N/m)	Vapor Pressure, p_v [N/m ² (abs)]	Bulk Modulus, ^b E_v (N/m ²)
Carbon tetrachloride	20	1,590	15.6	9.58 E − 4	6.03 E − 7	2.69 E − 2	1.3 E + 4	1.31 E + 9
Ethyl alcohol	20	789	7.74	1.19 E − 3	1.51 E − 6	2.28 E − 2	5.9 E + 3	1.06 E + 9
Gasoline ^c	15.6	680	6.67	3.1 E − 4	4.6 E − 7	2.2 E − 2	5.5 E + 4	1.3 E + 9
Glycerin	20	1,260	12.4	1.50 E + 0	1.19 E − 3	6.33 E − 2	1.4 E − 2	4.52 E + 9
Mercury	20	13,600	133	1.57 E − 3	1.15 E − 7	4.66 E − 1	1.6 E − 1	2.85 E + 10
SAE 30 oil ^c	15.6	912	8.95	3.8 E − 1	4.2 E − 4	3.6 E − 2	—	1.5 E + 9
Seawater	15.6	1,030	10.1	1.20 E − 3	1.17 E − 6	7.34 E − 2	1.77 E + 3	2.34 E + 9
Water	15.6	999	9.80	1.12 E − 3	1.12 E − 6	7.34 E − 2	1.77 E + 3	2.15 E + 9

^aIn contact with air.
^bIsentropic bulk modulus calculated from speed of sound.
^cTypical values. Properties of petroleum products vary.

Specific weight is weight per unit volume; specific gravity is the ratio of fluid density to the density of water at a certain temperature.



1.4.2 Specific Weight

The **specific weight** of a fluid, designated by the Greek symbol γ (gamma), is defined as its *weight* per unit volume. Thus, specific weight is related to density through the equation

$$\gamma = \rho g \quad (1.6)$$

where g is the local acceleration of gravity. Just as density is used to characterize the mass of a fluid system, the specific weight is used to characterize the weight of the system. In the BG and EE systems, γ has units of lb/ft^3 and in SI the units are N/m^3 . Under conditions of standard gravity ($g = 32.174 \text{ ft}/\text{s}^2 = 9.807 \text{ m}/\text{s}^2$), water at 60°F has a specific weight of $62.4 \text{ lb}/\text{ft}^3$ and $9.80 \text{ kN}/\text{m}^3$. Tables 1.5 and 1.6 list values of specific weight for several common liquids (based on standard gravity). More complete tables for water can be found in Appendix B (Tables B.1 and B.2).

1.4.3 Specific Gravity

The **specific gravity** of a fluid, designated as SG , is defined as the ratio of the density of the fluid to the density of water at some specified temperature. Usually the specified temperature is taken as 4°C (39.2°F), and at this temperature the density of water is $1.94 \text{ slugs}/\text{ft}^3$ or $1000 \text{ kg}/\text{m}^3$. In equation form, specific gravity is expressed as

$$SG = \frac{\rho}{\rho_{\text{H}_2\text{O}@4^\circ\text{C}}} \quad (1.7)$$

and since it is the *ratio* of densities, the value of SG does not depend on the system of units used. For example, the specific gravity of mercury at 20°C is 13.55. This is illustrated in the adjacent figure. Thus, the density of mercury can be readily calculated in either BG or SI units through the use of Eq. 1.7 as

$$\rho_{\text{Hg}} = (13.55)(1.94 \text{ slugs}/\text{ft}^3) = 26.3 \text{ slugs}/\text{ft}^3$$

or

$$\rho_{\text{Hg}} = (13.55)(1000 \text{ kg}/\text{m}^3) = 13.6 \times 10^3 \text{ kg}/\text{m}^3$$

It is clear that density, specific weight, and specific gravity are all interrelated, and from a knowledge of any one of the three the others can be calculated.

1.5 Ideal Gas Law

Compared to liquids, gases are highly compressible. The **ideal gas law** is frequently used as an *equation of state* to define a relationship between pressure, density, and temperature of a gas. In the study of thermodynamics, this relationship is typically expressed by the equation

$$p\nu = RT$$

where p is the absolute pressure, ν is the specific volume, R is the gas constant, and T is the absolute temperature.² In the study of fluid mechanics, we use density more often than specific volume so an alternative form of this equation more commonly encountered,

$$p = \rho RT \quad (1.8)$$

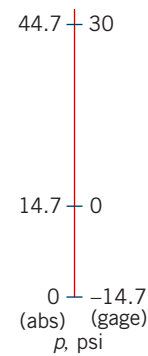
This relatively simple equation of state provides a good model for gases under typical conditions when they are not approaching liquefaction.

Pressure in a fluid at rest is defined as the normal force per unit area exerted on a plane surface (real or imaginary) immersed in a fluid and is created by the bombardment of the surface with the fluid molecules. From the definition, pressure has the dimension of FL^{-2} . In BG and EE units, pressure is expressed as lb/ft^2 (psf) or lb/in^2 (psi) and in SI units as N/m^2 . In SI, $1 \text{ N}/\text{m}^2$ is defined as a *pascal*, abbreviated as Pa, and pressures are commonly

In the ideal gas law, absolute pressures and temperatures must be used.

²We will use T to represent temperature in thermodynamic relationships, although T is also used to denote the basic dimension of time.

specified in pascals. The pressure in the ideal gas law must be expressed as an **absolute pressure**, denoted (abs), which means that it is measured relative to absolute zero pressure (a pressure that would only occur in a perfect vacuum). Standard sea-level atmospheric pressure (by international agreement) is 14.696 psi (abs) or 101.33 kPa (abs). For most calculations, these pressures can be rounded to 14.7 psi and 101 kPa, respectively. In engineering it is common practice to measure pressure relative to the local atmospheric pressure, and when measured in this fashion, it is called **gauge pressure**. Thus, the absolute pressure can be obtained from the gage pressure by adding the value of the atmospheric pressure. For example, as shown in the adjacent figure, a pressure of 30 psi (gage) in a tire is equal to 44.7 psi (abs) at standard atmospheric pressure. Pressure is a particularly important fluid characteristic, and it will be discussed more fully in the next chapter.



EXAMPLE 1.3 | Ideal Gas Law

Given The compressed air tank shown in Fig. E1.3a has a volume of 0.84 ft³. The temperature is 70 °F, and the atmospheric pressure is 14.7 psi (abs).

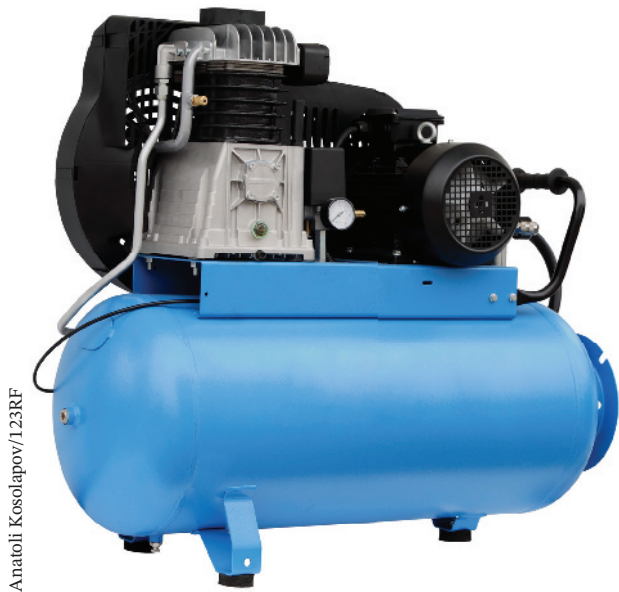


FIGURE E1.3a

Find When the tank is filled with air at a gage pressure of 50 psi, determine the density of the air and the weight of air in the tank.

Solution The air density can be obtained from the ideal gas law (Eq. 1.8)

$$\rho = \frac{p}{RT}$$

so that

$$\begin{aligned} \rho &= \frac{(50 \text{ lb/in.}^2 + 14.7 \text{ lb/in.}^2)(144 \text{ in.}^2/\text{ft}^2)}{(1716 \text{ ft} \cdot \text{lb/slug} \cdot ^\circ\text{R})(70 + 460) ^\circ\text{R}} \\ &= 0.0102 \text{ slugs/ft}^3 \end{aligned} \quad (\text{Ans})$$

Note that both the pressure and temperature were changed to absolute values.

The weight, $\mathcal{W}$, of the air is equal to

$$\begin{aligned} \mathcal{W} &= \rho g \times (\text{volume}) \\ &= (0.0102 \text{ slug/ft}^3)(32.2 \text{ ft/s}^2)(0.84 \text{ ft}^3) \\ &= 0.276 \text{ slug} \cdot \text{ft/s}^2 \end{aligned}$$

so that since 1 lb = 1 slug · ft/s²

$$\mathcal{W} = 0.276 \text{ lb} \quad (\text{Ans})$$

Comments By repeating the calculations for various values of the pressure, p , the results shown in Fig. E1.3b are obtained. Note that doubling the gage pressure does not double the amount of air in the tank, but doubling the absolute pressure does. Thus, a scuba diving tank at a gage pressure of 100 psi does not contain twice the amount of air as when the gage reads 50 psi.

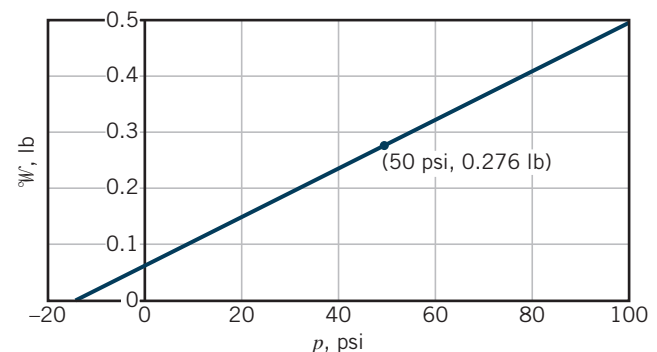


FIGURE E1.3b

The gas constant, R , which appears in Eq. 1.8, depends on the particular gas and is related to the molecular weight of the gas. Values of the gas constant for several common gases are listed in Tables 1.7 and 1.8. Also, in these tables, the gas density and specific weight are given for standard atmospheric pressure and gravity and for the temperature listed. More complete tables for air at standard atmospheric pressure can be found in Appendix B (Tables B.3 and B.4).

TABLE 1.7 Approximate Physical Properties of Some Common Gases at Standard Atmospheric Pressure (BG Units)

Gas	Temperature (°F)	Density, ρ (slugs/ft ³)	Specific Weight, γ (lb/ft ³)	Dynamic Viscosity, μ (lb · s/ft ²)	Kinematic Viscosity, ν (ft ² /s)	Gas Constant, ^a R (ft · lb/slug · °R)	Specific Heat Ratio, ^b k
Air (standard)	59	2.38 E − 3	7.65 E − 2	3.74 E − 7	1.57 E − 4	1.716 E + 3	1.40
Carbon dioxide	68	3.55 E − 3	1.14 E − 1	3.07 E − 7	8.65 E − 5	1.130 E + 3	1.30
Helium	68	3.23 E − 4	1.04 E − 2	4.09 E − 7	1.27 E − 3	1.242 E + 4	1.66
Hydrogen	68	1.63 E − 4	5.25 E − 3	1.85 E − 7	1.13 E − 3	2.466 E + 4	1.41
Methane (natural gas)	68	1.29 E − 3	4.15 E − 2	2.29 E − 7	1.78 E − 4	3.099 E + 3	1.31
Nitrogen	68	2.26 E − 3	7.28 E − 2	3.68 E − 7	1.63 E − 4	1.775 E + 3	1.40
Oxygen	68	2.58 E − 3	8.31 E − 2	4.25 E − 7	1.65 E − 4	1.554 E + 3	1.40

^aValues of the gas constant are independent of temperature.
^bValues of the specific heat ratio depend only slightly on temperature.

TABLE 1.8 Approximate Physical Properties of Some Common Gases at Standard Atmospheric Pressure (SI Units)

Gas	Temperature (°C)	Density, ρ (kg/m ³)	Specific Weight, γ (N/m ³)	Dynamic Viscosity, μ (N · s/m ²)	Kinematic Viscosity, ν (m ² /s)	Gas Constant, ^a R (1/kg · K)	Specific Heat Ratio, ^b k
Air (standard)	15	1.23 E + 0	1.20 E + 1	1.79 E − 5	1.46 E − 5	2.869 E + 2	1.40
Carbon dioxide	20	1.83 E + 0	1.80 E + 1	1.47 E − 5	8.03 E − 6	1.889 E + 2	1.30
Helium	20	1.66 E − 1	1.63 E + 0	1.94 E − 5	1.15 E − 4	2.077 E + 3	1.66
Hydrogen	20	8.38 E − 2	8.22 E − 1	8.84 E − 6	1.05 E − 4	4.124 E + 3	1.41
Methane (natural gas)	20	6.67 E − 1	6.54 E + 0	1.10 E − 5	1.65 E − 5	5.183 E + 2	1.31
Nitrogen	20	1.16 E + 0	1.14 E + 1	1.76 E − 5	1.52 E − 5	2.968 E + 2	1.40
Oxygen	20	1.33 E + 0	1.30 E + 1	2.04 E − 5	1.53 E − 5	2.598 E + 2	1.40

^aValues of the gas constant are independent of temperature.
^bValues of the specific heat ratio depend only slightly on temperature.

1.6 Viscosity

The properties of density and specific weight are measures of the “heaviness” of a fluid. It is clear, however, that these properties are not sufficient to uniquely characterize how fluids behave since two fluids (such as water and oil) can have approximately the same value of density but behave quite differently when flowing. Apparently, some additional property is needed to describe the “fluidity” of the fluid.

To determine this additional property, consider a hypothetical experiment in which a material is placed between two very wide parallel plates, as shown in **Fig. 1.4a**. The bottom plate is rigidly fixed, but the upper plate is free to move. If a solid, such as steel, were placed between the two plates and loaded with the force P as shown, the top plate would be displaced through some small distance, δa (assuming the solid was mechanically attached to the plates). The vertical line AB would be rotated through the small angle, $\delta\beta$, to the new position AB' . We note that to resist the applied force, P , a shearing stress, τ , would be developed at the plate–material interface, and for equilibrium to occur, $P = \tau A$ where A is the effective upper plate area (**Fig. 1.4b**). It is well-known that for elastic solids, such as steel, the small angular displacement, $\delta\beta$ (called the shearing strain), is proportional to the shearing stress, τ , that is developed in the material.

What happens if the solid is replaced with a fluid such as water? We would immediately notice a major difference. When the force P is applied to the upper plate, it will move continuously with a velocity, U (after the initial transient motion has died out), as illustrated in **Fig. 1.5**. This behavior is consistent with the definition of a fluid—that is, if a shearing stress is applied to a fluid, it will deform continuously. A closer inspection of the fluid motion between the two plates would reveal that the fluid in contact with the upper plate moves with the plate velocity, U , and the fluid in contact with the bottom fixed plate has a zero velocity. The fluid between the two plates moves with velocity $u = u(y)$ that would be found to vary linearly, $u = Uy/b$, as illustrated in **Fig. 1.5**. Thus, a *velocity gradient*, du/dy , is developed in the fluid between the plates. In this particular case, the velocity gradient is a constant because $du/dy = U/b$, but in more complex flow situations, such as that shown in the adjacent photograph, this is not true. The experimental observation that the fluid “sticks” to the solid boundaries is a very important one in fluid mechanics and is usually referred to as the **no-slip condition**. All fluids, both liquids and gases, satisfy this condition for typical flows.

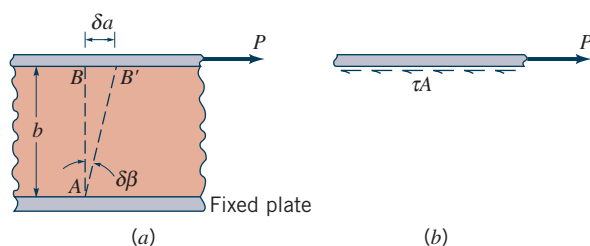


FIGURE 1.4 (a) Deformation of material placed between two parallel plates. (b) Forces acting on upper plate.

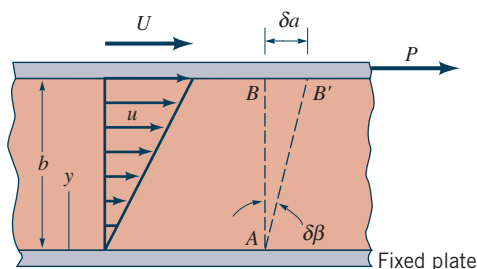


FIGURE 1.5 Behavior of a fluid placed between two parallel plates.

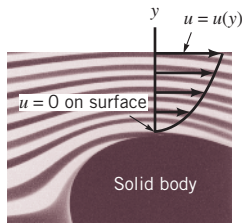


V1.3 Viscous fluids



V1.4 No-slip condition

Real fluids, even though they may be moving, always “stick” to the solid boundaries that contain them.



In a small time increment, δt , an imaginary vertical line AB in the fluid would rotate through an angle, $\delta\beta$, so that

$$\delta\beta \approx \tan \delta\beta = \frac{\delta a}{b}$$

Since $\delta a = U \delta t$, it follows that

$$\delta\beta = \frac{U \delta t}{b}$$

We note that in this case, $\delta\beta$ is a function not only of the force P (which governs U) but also of time. Thus, it is not reasonable to attempt to relate the shearing stress, τ , to $\delta\beta$ as is done for solids. Rather, we consider the *rate* at which $\delta\beta$ is changing and define the **rate of shearing strain**, $\dot{\gamma}$, as

$$\dot{\gamma} = \lim_{\delta t \rightarrow 0} \frac{\delta\beta}{\delta t}$$

which in this instance is equal to

$$\dot{\gamma} = \frac{U}{b} = \frac{du}{dy}$$

For many common fluids, continuation of this experiment would reveal that as the shearing stress, τ , is increased by increasing P (recall that $\tau = P/A$), the rate of shearing strain is increased in direct proportion—that is,

$$\tau \propto \dot{\gamma}$$

or

$$\tau \propto \frac{du}{dy}$$

This result indicates that for common fluids such as water, oil, gasoline, and air

$$\tau = \mu \frac{du}{dy} \quad (1.9)$$

where the constant of proportionality between the applied shear stress and the resulting strain rate is designated by the Greek symbol μ (mu) and is called the **absolute viscosity**, *dynamic viscosity*, or simply the *viscosity* of the fluid. In accordance with Eq. 1.9, plots of τ vs. du/dy should be linear with the slope equal to the viscosity, as illustrated in **Fig. 1.6**. The actual value



Video

V1.5 Capillary tube viscometer

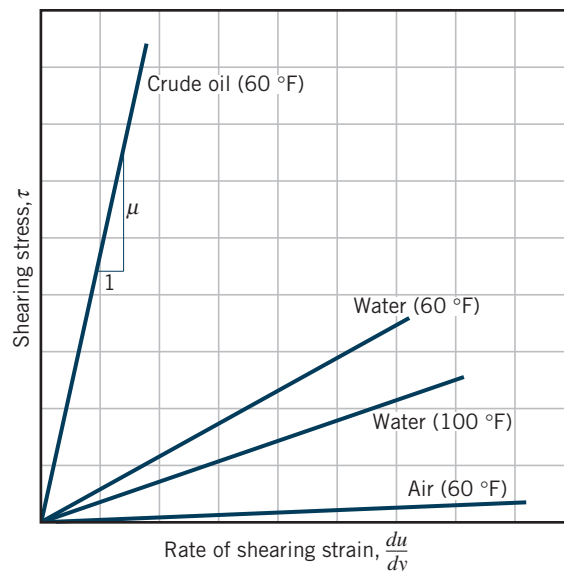


FIGURE 1.6 Linear variation of shearing stress with rate of shearing strain for common fluids.

of the viscosity depends on the particular fluid, and for a particular fluid, the viscosity is also highly dependent on temperature, as illustrated in Fig. 1.6, with the two curves for water. Fluids for which the shearing stress is *linearly* related to the rate of shearing strain (also referred to as the rate of angular deformation) are designated as **Newtonian fluids** after Isaac Newton (1642–1727). Fortunately, most common fluids, both liquids and gases, are Newtonian. A more general formulation of Eq. 1.9, which applies to more complex flows of Newtonian fluids, is given in Section 6.8.1.

Dynamic viscosity is the fluid property that relates shearing stress and fluid motion.

The Wide World of Fluids

An extremely viscous fluid

Pitch is a derivative of tar once used for waterproofing boats. At elevated temperatures it flows quite readily. At room temperature it feels like a solid—it can even be shattered with a blow from a hammer. However, it is a liquid. In 1927, Professor Parnell heated some pitch and poured it into a funnel. Since that time, it has been

allowed to flow freely (or rather, drip slowly) from the funnel. The flowrate is quite small. In fact, to date only seven drops have fallen from the end of the funnel, although the eighth drop is poised ready to fall “soon.” While nobody has actually seen a drop fall from the end of the funnel, a beaker below the funnel holds the previous drops that fell over the years. It is estimated that the pitch is about 100 billion times more viscous than water.

Fluids for which the shearing stress is not linearly related to the rate of shearing strain are designated as **non-Newtonian fluids**. Although there is a variety of types of non-Newtonian fluids, the simplest and most common are shown in Fig. 1.7. The slope of the shearing stress vs. rate of shearing strain graph is denoted as the *apparent viscosity*, μ_{ap} . For Newtonian fluids, the apparent viscosity is the same as the viscosity and is independent of shear rate.

For *shear thinning fluids*, the apparent viscosity decreases with increasing shear rate—the harder the fluid is sheared, the less viscous it becomes. Many colloidal suspensions and polymer solutions are shear thinning. For example, latex paint does not drip from the brush because the shear rate is small and the apparent viscosity is large. However, it flows smoothly onto the wall because the thin layer of paint between the wall and the brush causes a large shear rate and a small apparent viscosity.

For *shear thickening fluids* the apparent viscosity increases with increasing shear rate—the harder the fluid is sheared, the more viscous it becomes. Common examples of this type of fluid include water–corn starch mixture and water–sand mixture (“quicksand”). Thus, the difficulty in removing an object from quicksand increases dramatically as the speed of removal increases.

The other type of behavior indicated in Fig. 1.7 is that of a *Bingham plastic*, which is neither a fluid nor a solid. Such material can withstand a finite, nonzero shear stress, τ_{yield} , the yield stress, without motion (therefore, it is not a fluid), but once the yield stress is exceeded it flows like a fluid (hence, it is not a solid). Toothpaste and mayonnaise are common examples

For non-Newtonian fluids, the apparent viscosity is a function of the shear rate.

The various types of non-Newtonian fluids are distinguished by how their apparent viscosity changes with shear rate.

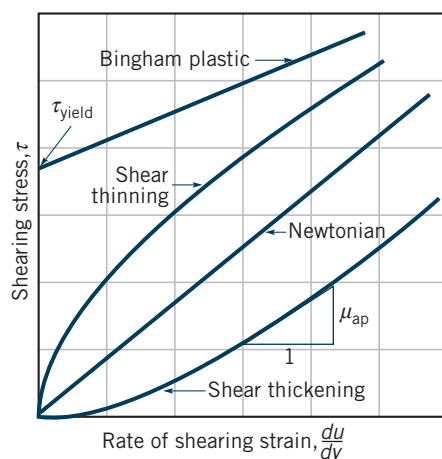


FIGURE 1.7 Variation of shearing stress with rate of shearing strain for several types of fluids, including common non-Newtonian fluids.

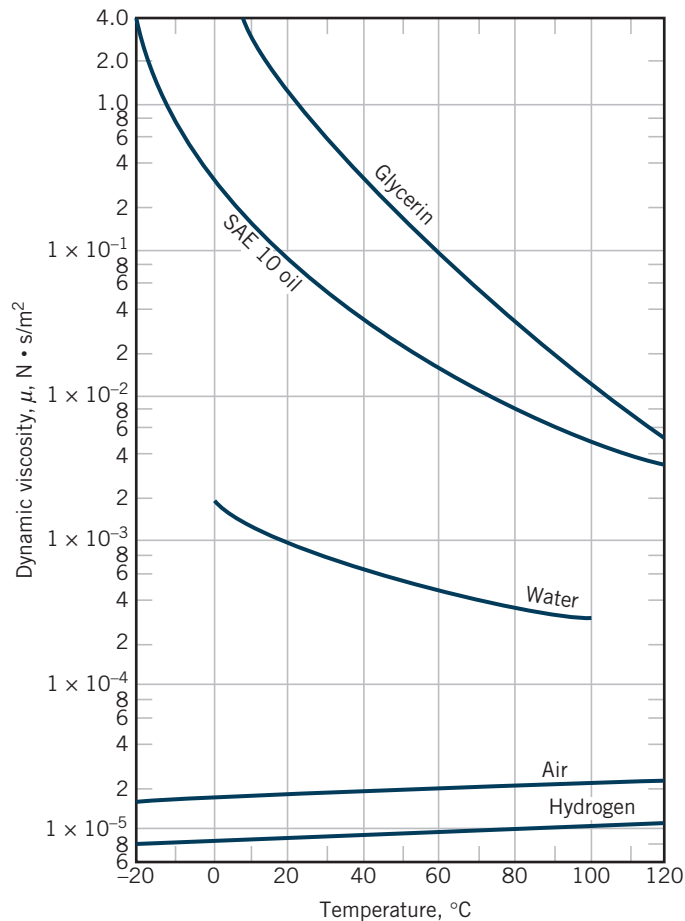
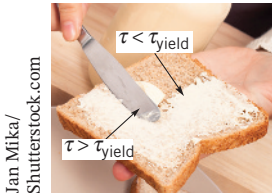


FIGURE 1.8 Dynamic (absolute) viscosity of some common fluids as a function of temperature.



of Bingham plastic materials. As indicated in the adjacent figure, mayonnaise can sit in a pile on a slice of bread (the shear stress less than the yield stress), but it flows smoothly into a thin layer when the knife increases the stress above the yield stress.

From Eq. 1.9 it can be readily deduced that the dimensions of viscosity are FTL^{-2} . Thus, in BG and EE units, viscosity is given as $lb \cdot s/ft^2$ and in SI units as $N \cdot s/m^2$. Values of viscosity for several common liquids and gases are listed in Tables 1.5 through 1.8. A quick glance at these tables reveals the wide variation in viscosity among fluids. Viscosity is only mildly dependent on pressure, and the influence of pressure on the value of viscosity is usually neglected. However, as previously mentioned, and as illustrated in **Fig. 1.8**, viscosity is very sensitive to temperature. For example, as the temperature of water changes from 60 to 100 °F, the density decreases by less than 1%, but the viscosity decreases by about 40%. It is thus clear that particular attention must be given to temperature when determining viscosity.

Figure 1.8 shows in more detail how the viscosity varies from fluid to fluid and how for a given fluid it varies with temperature. It is to be noted from this figure that the viscosity of liquids decreases with an increase in temperature, whereas for gases an increase in temperature causes an increase in viscosity. This difference in the effect of temperature on the viscosity of liquids and gases can again be traced back to the difference in molecular structure. The liquid molecules are closely spaced, with strong cohesive forces between molecules, and the resistance to relative motion between adjacent layers of fluid is related to these intermolecular forces. As the temperature increases, these cohesive forces are reduced with a corresponding reduction in resistance to motion. Since viscosity is an index of this resistance, it follows that the viscosity is reduced by an increase in temperature. In gases, however, the molecules are widely spaced and intermolecular forces negligible. In this case, resistance to relative motion arises due to the exchange of momentum of gas molecules between adjacent layers. As molecules are transported by random motion from a region of low bulk velocity to mix with molecules in a region of higher bulk velocity (and vice versa), there is an effective momentum



Video

V1.6 Non-Newtonian behavior

exchange that resists relative motion between the layers. As the temperature of the gas increases, the random molecular activity increases with a corresponding increase in viscosity.

The effect of temperature on viscosity can be closely approximated using two empirical formulas. For gases, the *Sutherland equation* can be expressed as

$$\mu = \frac{CT^{3/2}}{T + S} \quad (1.10)$$

Viscosity is very sensitive to temperature.

where C and S are empirical constants, and T is absolute temperature. Thus, if the viscosity is known at two temperatures, C and S can be determined. Or, if more than two viscosities are known, the data can be correlated with Eq. 1.10 by using some type of curve-fitting scheme.

For liquids, an empirical equation that has been used is

$$\mu = De^{B/T} \quad (1.11)$$

where D and B are constants and T is absolute temperature. This equation is often referred to as *Andrade's equation*. As was the case for gases, the viscosity must be known at least for two temperatures so the two constants can be determined. A more detailed discussion of the effect of temperature on fluids can be found in Ref. 1.

EXAMPLE 1.4 | Viscosity and Dimensionless Quantities

Given A dimensionless combination of variables that is important in the study of viscous flow through pipes is called the *Reynolds number*, Re , defined as $\rho VD/\mu$ where, as indicated in **Fig. E1.4**, ρ is the fluid density, V the mean fluid velocity, D the pipe diameter, and μ the fluid viscosity. A Newtonian fluid having a viscosity of $0.38 \text{ N} \cdot \text{s}/\text{m}^2$ and a specific gravity of 0.91 flows through a 25-mm-diameter pipe with a velocity of 2.6 m/s.

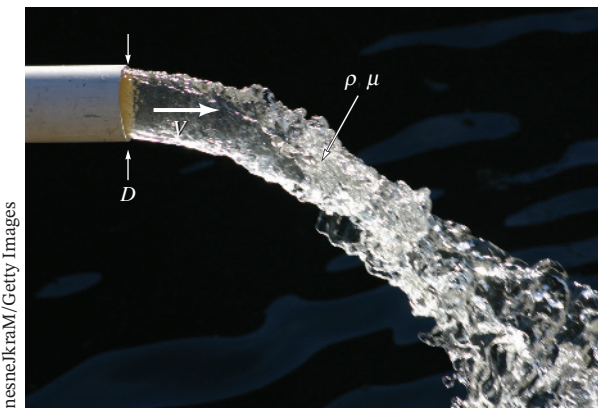


FIGURE E1.4

Find Determine the value of the Reynolds number using (a) SI units and (b) BG units.

Solution

- a. The fluid density is calculated from the specific gravity as

$$\rho = SG\rho_{\text{H}_2\text{O}@4^\circ\text{C}} = 0.91(1000 \text{ kg}/\text{m}^3) = 910 \text{ kg}/\text{m}^3$$

and from the definition of the Reynolds number

$$\begin{aligned} Re &= \frac{\rho VD}{\mu} = \frac{(910 \text{ kg}/\text{m}^3)(2.6 \text{ m}/\text{s})(25 \text{ mm})(10^{-3} \text{ m}/\text{mm})}{0.38 \text{ N} \cdot \text{s}/\text{m}^2} \\ &= 156 (\text{kg} \cdot \text{m}/\text{s}^2)/\text{N} \end{aligned}$$

However, since $1 \text{ N} = 1 \text{ kg} \cdot \text{m}/\text{s}^2$, it follows that the Reynolds number is unitless—that is,

$$Re = 156 \quad (\text{Ans})$$

The value of any dimensionless quantity does not depend on the system of units used if all variables that make up the quantity are expressed in a consistent set of units. To check this, we will calculate the Reynolds number using BG units.

- b. We first convert all the SI values of the variables appearing in the Reynolds number to BG values. Thus,

$$\rho = (910 \text{ kg}/\text{m}^3)(1.940 \times 10^{-3}) = 1.77 \text{ slugs}/\text{ft}^3$$

$$V = (2.6 \text{ m}/\text{s})(3.281) = 8.53 \text{ ft}/\text{s}$$

$$D = (0.025 \text{ m})(3.281) = 8.20 \times 10^{-2} \text{ ft}$$

$$\mu = (0.38 \text{ N} \cdot \text{s}/\text{m}^2)(2.089 \times 10^{-2}) = 7.94 \times 10^{-3} \text{ lb} \cdot \text{s}/\text{ft}^2$$

and the value of the Reynolds number is

$$\begin{aligned} Re &= \frac{(1.77 \text{ slugs}/\text{ft}^3)(8.53 \text{ ft}/\text{s})(8.20 \times 10^{-2} \text{ ft})}{7.94 \times 10^{-3} \text{ lb} \cdot \text{s}/\text{ft}^2} \\ &= 156 (\text{slug} \cdot \text{ft}/\text{s}^2)/\text{lb} = 156 \quad (\text{Ans}) \end{aligned}$$

since $1 \text{ lb} = 1 \text{ slug} \cdot \text{ft}/\text{s}^2$.

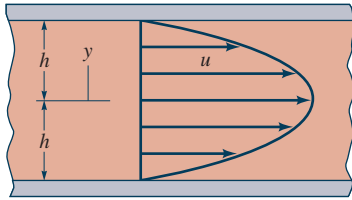
Comments The values from part (a) and part (b) are the same, as expected. Dimensionless quantities play an important role in fluid mechanics, and the significance of the Reynolds number as well as other important dimensionless combinations will be discussed in detail in Chapter 7. It should be noted that in the Reynolds number it is actually the ratio μ/ρ that is important, and this is the property that is defined as the kinematic viscosity.

EXAMPLE 1.5 | Newtonian Fluid Shear Stress

Given The velocity distribution for the flow of a Newtonian fluid between two fixed wide, parallel plates (see **Fig. E1.5a**) is given by the equation

$$u = \frac{3V}{2} \left[1 - \left(\frac{y}{h} \right)^2 \right]$$

where V is the mean velocity. The fluid has a viscosity of $0.04 \text{ lb} \cdot \text{s}/\text{ft}^2$. Also, $V = 2 \text{ ft/s}$ and $h = 0.2 \text{ in.}$

**FIGURE E1.5a**

Find Determine: (a) the shearing stress acting on the bottom wall, and (b) the shearing stress acting on a plane parallel to the walls and passing through the centerline (midplane).

Solution For this type of parallel flow, the shearing stress is obtained from Eq. 1.9,

$$\tau = \mu \frac{du}{dy} \quad (1)$$

Thus, if the velocity distribution $u = u(y)$ is known, the shearing stress can be determined at all points by evaluating the velocity gradient, du/dy . For the distribution given

$$\frac{du}{dy} = -\frac{3Vy}{h^2} \quad (2)$$

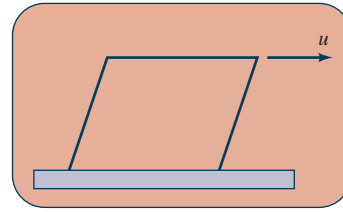
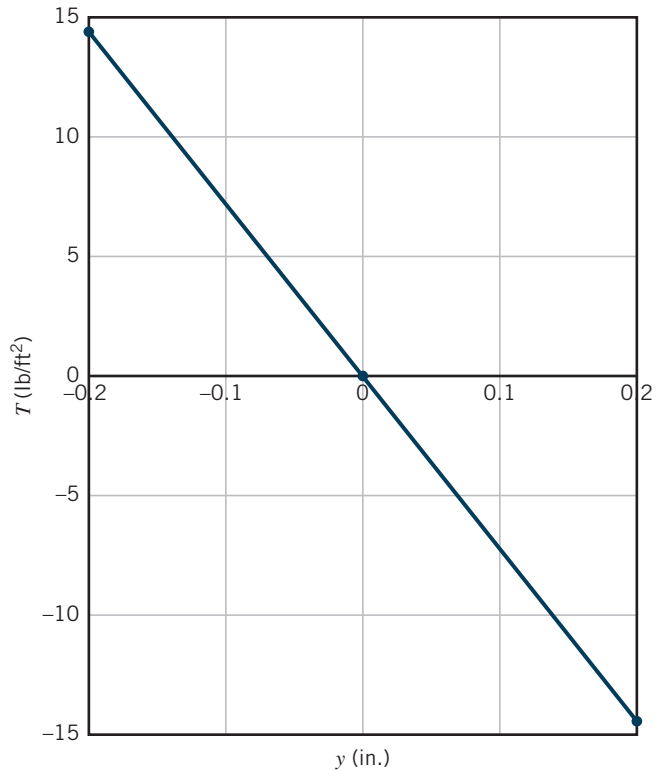
- a. Along the bottom wall $y = -h$ so that (from Eq. 2)

$$\frac{du}{dy} = \frac{3V}{h}$$

and therefore the shearing stress is

$$\begin{aligned} \tau_{\text{bottom wall}} &= \mu \left(\frac{3V}{h} \right) = \frac{(0.04 \text{ lb} \cdot \text{s}/\text{ft}^2)(3)(2 \text{ ft/s})}{(0.2 \text{ in.})(1 \text{ ft}/12 \text{ in.})} \\ &= 14.4 \text{ lb}/\text{ft}^2 \text{ (in direction of flow)} \quad (\text{Ans}) \end{aligned}$$

A positive shear stress seems intuitively satisfying. To check your intuition, sketch a sheared fluid particle adjacent to the bottom plate (**Fig. E1.5b**). The fluid motion seems to be trying to drag the plate in the flow direction. Reference to section 1.6 in which the concept of viscosity was introduced, and Fig. 1.4 in particular, makes clear that the shear stress on the fluid at the wall is indeed positive. A review of Eq. 2 and the answers to parts a and b of this example makes it clear that the shear stress varies linearly across the flow field.

**FIGURE E1.5b****FIGURE E1.5c**

- b. Along the midplane where $y = 0$ it follows from Eq. 2 that

$$\frac{du}{dy} = 0$$

and thus the shearing stress is

$$\tau_{\text{midplane}} = 0 \quad (\text{Ans})$$

Comment Equation 2 has been evaluated at the bottom plate and midplane to provide the answers to parts a and b of this example. If you use it to compute the shear stress at the top plate, you will notice that it produces a shear stress of the same magnitude as at the bottom plate, but of opposite sign. A plot of τ vs. y between the plates, **Fig. E1.5c**, confirms both the difference in sign and the earlier observation of the linear relationship. Can you explain the change of sign? *Hint:* Sketch a sheared fluid particle adjacent to the top plate.

Quite often viscosity appears in fluid flow problems combined with the density in the form

$$\nu = \frac{\mu}{\rho}$$

This ratio is called the **kinematic viscosity** and is denoted with the Greek symbol ν (nu). The dimensions of kinematic viscosity are L^2/T , and the BG units are ft^2/s and SI units are m^2/s . Values of kinematic viscosity for some common liquids and gases are given in Tables 1.5 through 1.8. More extensive tables giving both the dynamic and kinematic viscosities for water and air can be found in Appendix B (Tables B.1 through B.4), and graphs showing the variation in both dynamic and kinematic viscosity with temperature for a variety of fluids are also provided in Appendix B (Figs. B.1 and B.2).

Although in this text we are primarily using BG and SI units, dynamic viscosity is often expressed in the metric CGS (centimeter-gram-second) system with units of $\text{dyne} \cdot \text{s}/\text{cm}^2$. This combination is called a *poise*, abbreviated P. In the CGS system, kinematic viscosity has units of cm^2/s , and this combination is called a *stoke*, abbreviated St.

Kinematic viscosity is defined as the ratio of the absolute viscosity to the fluid density.

1.7 Compressibility of Fluids

1.7.1 Bulk Modulus

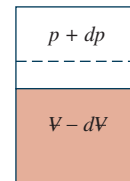
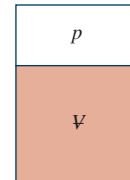
An important question to answer when considering the behavior of a particular fluid is how easily can the volume (and thus the density) of a given mass of the fluid be changed when there is a change in pressure? That is, how compressible is the fluid? A property that is commonly used to characterize compressibility is the **bulk modulus**, E_v , defined as

$$E_v = -\frac{dp}{d\mathcal{V}/\mathcal{V}} \quad (1.12)$$

where dp is the differential change in pressure needed to create a differential change in volume, $d\mathcal{V}$, of a volume $\mathcal{V}$. This is illustrated in the adjacent figure. The negative sign is included since an increase in pressure will cause a decrease in volume. Since a decrease in volume of a given mass, $m = \rho\mathcal{V}$, will result in an increase in density, Eq. 1.12 can also be expressed as

$$E_v = \frac{dp}{d\rho/\rho} \quad (1.13)$$

The bulk modulus (also referred to as the *bulk modulus of elasticity*) has dimensions of pressure, FL^{-2} . In BG units, values for E_v are usually given as lb/in^2 (psi) and in SI units as N/m^2 (Pa). Large values for the bulk modulus indicate that the fluid is relatively incompressible—that is, it takes a large pressure change to create a small change in volume. As expected, values of E_v for common liquids are large (see Tables 1.5 and 1.6). For example, at atmospheric pressure and a temperature of 60°F , it would require a pressure of 3120 psi to compress a volume of water 1%. This result is representative of the compressibility of liquids. Since such large pressures are required to effect a change in volume, we conclude that liquids can be considered as *incompressible* for most practical engineering applications. As liquids are compressed the bulk modulus increases, but the bulk modulus near atmospheric pressure is usually the one of interest. The use of bulk modulus as a property describing compressibility is most prevalent when dealing with liquids, although the bulk modulus can also be determined for gases.



Video

V1.7 Water balloon

The Wide World of Fluids

This water jet is a blast

Usually liquids can be treated as incompressible fluids. However, in some applications the *compressibility* of a liquid can play a key role in the operation of a device. For example, a water pulse generator using compressed water has been developed for use in mining operations. It can fracture rock by producing an effect comparable to a conventional explosive such as gunpowder. The device uses the energy stored in a water-filled accumulator to generate an ultrahigh-pressure water pulse ejected through a 10- to 25-mm-diameter

discharge valve. At the ultrahigh pressures used (300 to 400 MPa, or 3000 to 4000 atmospheres), the water is compressed (i.e., the volume reduced) by about 10 to 15%. When a fast-opening valve within the pressure vessel is opened, the water expands and produces a jet of water that upon impact with the target material produces an effect similar to the explosive force from conventional explosives. Mining with the water jet can eliminate various hazards that arise with the use of conventional chemical explosives, such as those associated with the storage and use of explosives and the generation of toxic gas by-products that require extensive ventilation (see Problem 1.110).

1.7.2 Compression and Expansion of Gases

When gases are compressed (or expanded), the relationship between pressure and density depends on the nature of the process. If the compression or expansion takes place under constant temperature conditions (*isothermal process*), then from Eq. 1.8

$$\frac{p}{\rho} = \text{constant} \quad (1.14)$$

If the compression or expansion is frictionless and no heat is exchanged with the surroundings (*isentropic process*), then

$$\frac{p}{\rho^k} = \text{constant} \quad (1.15)$$

where k is the ratio of the specific heat at constant pressure, c_p , to the specific heat at constant volume, c_v (i.e., $k = c_p/c_v$). The two specific heats are related to the gas constant, R , through the equation $R = c_p - c_v$. As was the case for the ideal gas law, the pressure in both Eqs. 1.14 and 1.15 must be expressed as an absolute pressure. Values of k for some common gases are given in Tables 1.7 and 1.8 and, for air over a range of temperatures, in Appendix B (Tables B.3 and B.4). The pressure-density variations for isothermal and isentropic conditions are illustrated in the adjacent figure.

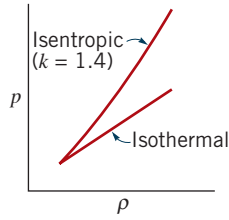
With explicit equations relating pressure and density, the bulk modulus for gases can be determined by obtaining the derivative $dp/d\rho$ from Eq. 1.14 or 1.15 and substituting the results into Eq. 1.13. It follows that for an isothermal process

$$E_v = p \quad (1.16)$$

and for an isentropic process,

$$E_v = kp \quad (1.17)$$

Note that in both cases the bulk modulus varies directly with pressure. For air under standard atmospheric conditions with $p = 14.7$ psi (abs) and $k = 1.40$, the isentropic bulk modulus is 20.6 psi. A comparison of this figure with that for water under the same conditions ($E_v = 312,000$ psi) shows that air is approximately 15,000 times as compressible as water. It is thus clear that in dealing with gases, greater attention will need to be given to the effect of compressibility on fluid behavior. However, as will be discussed further in later sections, gas flows can often be treated as incompressible flows if the changes in pressure are small.



The value of the bulk modulus depends on the type of process involved.

EXAMPLE 1.6 | Isentropic Compression of a Gas

Given A cubic foot of air at an absolute pressure of 14.7 psi is compressed isentropically to $\frac{1}{2}$ ft³ by the tire pump shown in Fig. E1.6a.

Find What is the final pressure?

Solution Assuming the compression occurs so quickly that the effect of heat transfer to the surroundings can be neglected, we use the relationship for isentropic compression

$$\frac{p_i}{\rho_i^k} = \frac{p_f}{\rho_f^k}$$

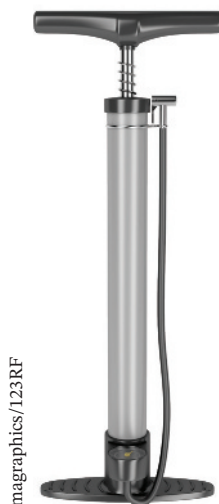
where the subscripts i and f refer to initial and final states, respectively. Since we are interested in the final pressure, p_f , it follows that

$$p_f = \left(\frac{\rho_f}{\rho_i}\right)^k p_i$$

As the volume, $\mathcal{V}$, is reduced by one-half, the density must double, since the mass, $m = \rho\mathcal{V}$, of the gas remains constant. Thus, with $k = 1.40$ for air

$$p_f = (2)^{1.40}(14.7 \text{ psi}) = 38.8 \text{ psi (abs)} \quad (\text{Ans})$$

Comment By repeating the calculations for various values of the ratio of the final volume to the initial volume, V_f/V_i , the results shown in Fig. E1.6b are obtained. Note that even though



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FIGURE E1.6a

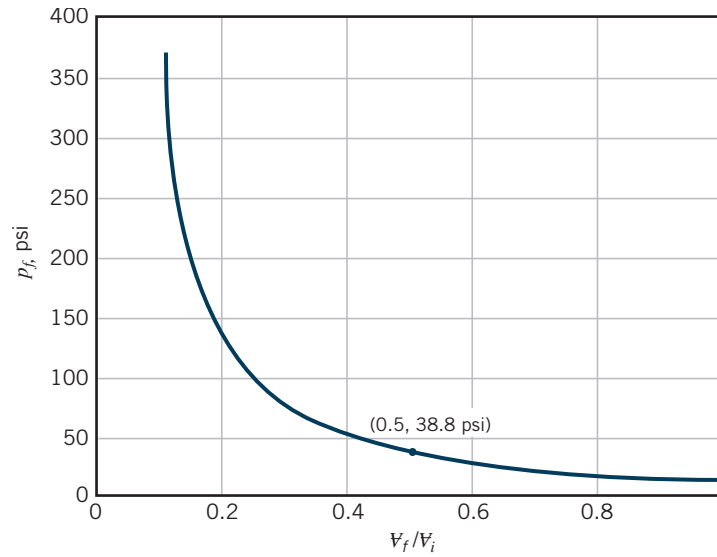


FIGURE E1.6b

air is often considered to be easily compressed (at least compared to liquids), it takes considerable pressure to significantly reduce a given volume of air as is done in an automobile engine where the compression ratio is on the order of $V_f/V_i = 1/8 = 0.125$. If the compression is so slow that the isothermal

model is appropriate, would the same change in pressure be required to accomplish the specified compression? What would a plot of pressure vs. compression ratio look like for an isothermal compression? Which assumption do you think is a better model for the process in an automobile engine?

1.7.3 Speed of Sound

Another important consequence of the compressibility of fluids is that disturbances introduced at some point in the fluid propagate at a finite velocity. For example, if a fluid is flowing in a pipe and a valve at the outlet is suddenly closed (thereby creating a localized disturbance), the effect of the valve closure is not felt instantaneously upstream. It takes a finite time for the increased pressure created by the valve closure to propagate to an upstream location. Similarly, a loudspeaker diaphragm causes a localized disturbance as it vibrates, and the small change in pressure created by the motion of the diaphragm is propagated through the air with a finite velocity. The velocity at which an infinitesimal disturbance propagates is called the *acoustic velocity* or the **speed of sound**, c . It will be shown in Chapter 11 that the speed of sound is related to changes in pressure and density of the fluid medium through the equation

$$c = \sqrt{\frac{dp}{d\rho}} \quad (1.18)$$

or in terms of the bulk modulus defined by Eq. 1.13

$$c = \sqrt{\frac{E_v}{\rho}} \quad (1.19)$$

Since the disturbance is small, there is negligible heat transfer and the process is assumed to be isentropic. Thus, the pressure–density relationship used in Eq. 1.18 is that for an isentropic process.

For gases undergoing an isentropic process, $E_v = kp$ (Eq. 1.17) so that

$$c = \sqrt{\frac{kp}{\rho}}$$

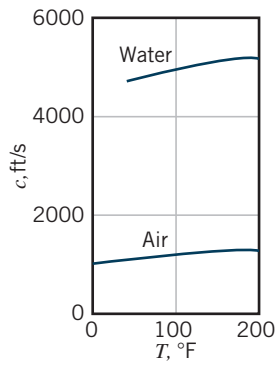
and making use of the ideal gas law, it follows that

$$c = \sqrt{kRT} \quad (1.20)$$

The velocity at which small disturbances propagate in a fluid is called the speed of sound.



V1.8 As fast as a speeding bullet



Thus, for ideal gases, the speed of sound is proportional to the square root of the absolute temperature. For example, for air at 60 °F with $k = 1.40$ and $R = 1716 \text{ ft} \cdot \text{lb}/\text{slug} \cdot ^\circ\text{R}$, it follows that $c = 1117 \text{ ft/s}$. The speed of sound in air at various temperatures can be found in Appendix B (Tables B.3 and B.4). Equation 1.19 is also valid for liquids, and values of E_v can be used to determine the speed of sound in liquids. For water at 20 °C, $E_v = 2.19 \text{ GN/m}^2$ and $\rho = 998.2 \text{ kg/m}^3$ so that $c = 1481 \text{ m/s}$ or 4860 ft/s . As shown in the adjacent figure, the speed of sound is much higher in water than in air. If a fluid were truly incompressible ($E_v = \infty$), the speed of sound would be infinite. The speed of sound in water for various temperatures can be found in Appendix B (Tables B.1 and B.2).

EXAMPLE 1.7 | Speed of Sound and Mach Number

Given A jet aircraft flies at a speed of 550 mph at an altitude of 35,000 ft, where the temperature is -66°F and the specific heat ratio is $k = 1.4$.

Find Determine the ratio of the speed of the aircraft, V , to that of the speed of sound, c , at the specified altitude.

Solution From Eq. 1.20, the speed of sound can be calculated as

$$\begin{aligned} c &= \sqrt{kRT} \\ &= \sqrt{(1.40)(1716 \text{ ft} \cdot \text{lb}/\text{slug} \cdot ^\circ\text{R})(-66 + 460)^\circ\text{R}} \\ &= 973 \text{ ft/s} \end{aligned}$$

Since the air speed is

$$V = \frac{(550 \text{ mi/hr})(5280 \text{ ft/mi})}{(3600 \text{ s/hr})} = 807 \text{ ft/s}$$

the ratio is

$$\frac{V}{c} = \frac{807 \text{ ft/s}}{973 \text{ ft/s}} = 0.829 \quad (\text{Ans})$$

Comment This ratio is called the *Mach number*, Ma . If $\text{Ma} < 1.0$, the aircraft is flying at *subsonic* speeds, whereas for $\text{Ma} > 1.0$ it is flying at *supersonic* speeds. The Mach number is an important dimensionless parameter used in the study of the flow of gases at high speeds and will be further discussed in Chapters 7 and 11.

By repeating the calculations for different temperatures, the results shown in **Fig. E1.7** are obtained. Because the speed of sound increases with increasing temperature, for a constant airplane speed, the Mach number decreases as the temperature increases.

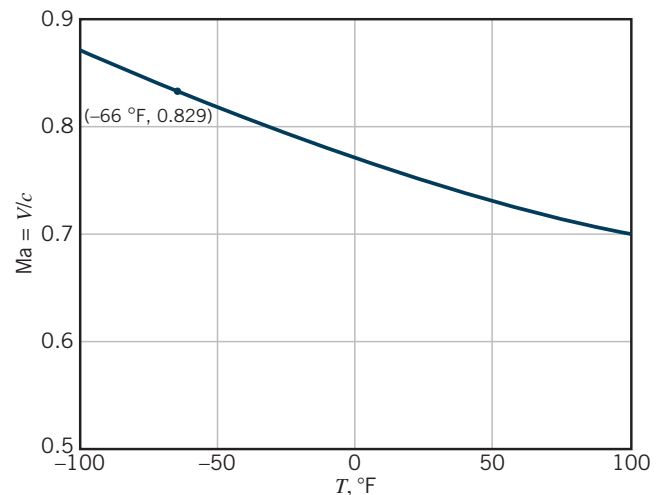
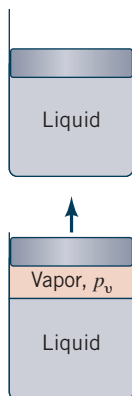


FIGURE E1.7

1.8 Vapor Pressure



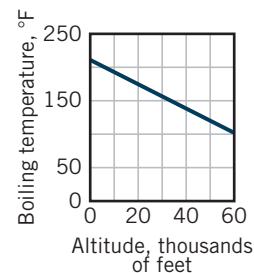
A liquid boils when the pressure is reduced to the vapor pressure.

It is a common observation that liquids such as water and gasoline will evaporate if they are simply placed in a container open to the atmosphere. Evaporation takes place because some liquid molecules at the surface have sufficient momentum to overcome the intermolecular cohesive forces and escape into the atmosphere. If the container is closed with a small air space left above the surface, and this space evacuated to form a vacuum, a pressure will develop in the space as a result of the vapor that is formed by the escaping molecules. When an equilibrium condition is reached so that the number of molecules leaving the surface is equal to the number entering, the vapor is said to be saturated and the pressure that the vapor exerts on the liquid surface is termed the **vapor pressure**, p_v . Similarly, if the end of a completely liquid-filled container is moved, as shown in the adjacent figure, without letting any air into the container, the space between the liquid and the end becomes filled with vapor at a pressure equal to the vapor pressure.

Since the development of a vapor pressure is closely associated with molecular activity, the value of vapor pressure for a particular liquid depends on temperature. Values of vapor pressure for water at various temperatures can be found in Appendix B (Tables B.1 and B.2), and the values of vapor pressure for several common liquids at room temperatures are given in Tables 1.5 and 1.6.

Boiling, which is the formation of vapor bubbles within a fluid mass, is initiated when the absolute pressure in the fluid reaches the vapor pressure. As commonly observed in the kitchen, water at standard atmospheric pressure will boil when the temperature reaches 212 °F (100 °C)—that is, the vapor pressure of water at 212 °F is 14.7 psi (abs). However, if we attempt to boil water at a higher elevation, say 30,000 ft above sea level (the approximate elevation of Mt. Everest), where the atmospheric pressure is 4.37 psi (abs), we find that boiling will start when the temperature is about 157 °F. At this temperature the vapor pressure of water is 4.37 psi (abs). For the U.S. Standard Atmosphere (see Section 2.4), the boiling temperature is a function of altitude as shown in the adjacent figure. Thus, boiling can be induced at a given pressure acting on the fluid by raising the temperature, or at a given fluid temperature by lowering the pressure.

An important reason for our interest in vapor pressure and boiling lies in the common observation that in flowing fluids it is possible to develop very low pressure due to the fluid motion, and if the pressure is lowered to the vapor pressure, boiling will occur. For example, this phenomenon may occur in flow through the irregular, narrowed passages of a valve or pump. When vapor bubbles are formed in a flowing fluid, they are swept along into regions of higher pressure where they suddenly collapse with sufficient intensity to actually cause structural damage. The formation and subsequent collapse of vapor bubbles in a flowing fluid, called *cavitation*, is an important fluid flow phenomenon to be given further attention in Chapters 3, 7, and 12.



In flowing liquids it is possible for the pressure in localized regions to reach vapor pressure, thereby causing cavitation.

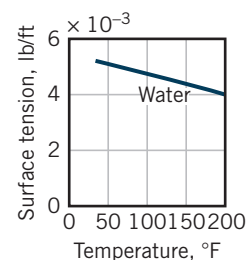
1.9 Surface Tension

At the interface between a liquid and a gas, or between two immiscible liquids, forces develop in the liquid surface that cause the surface to behave as if it were a “skin” or “membrane” stretched over the fluid mass. Although such a skin is not actually present, this conceptual analogy allows us to explain several commonly observed phenomena. For example, a steel needle or a razor blade will float on water if placed gently on the surface because the tension developed in the hypothetical skin supports it. Small droplets of mercury will form into spheres when placed on a smooth surface because the cohesive forces in the surface of the mercury tend to hold all the molecules together in a compact shape.

These various types of surface phenomena are due to the unbalanced cohesive forces acting on the liquid molecules at the fluid surface. Molecules in the interior of the fluid mass are surrounded by molecules that are attracted to each other equally. However, molecules along the surface may experience a net force toward the interior. The apparent physical consequence of this unbalanced force along the surface is to create the hypothetical skin or membrane. A tensile force may be considered to be acting in the plane of the surface along any line in the surface. The intensity of the molecular attraction per unit length along any line in the surface is called the **surface tension** and is designated by the Greek symbol σ (sigma). For a given liquid the surface tension depends on temperature as well as the other fluid it is in contact with at the interface. The dimensions of surface tension are FL^{-1} with BG units of lb/ft and SI units of N/m. Values of surface tension for some common liquids (in contact with air) are given in



V1.9 Floating razor blade



The Wide World of Fluids

Walking on water

Water striders are insects commonly found on ponds, rivers, and lakes that appear to “walk” on water. A typical length of a water strider is about 0.4 in., and they can cover 100 body lengths in one second. It has long been recognized that it is *surface tension* that keeps the water strider from sinking below the surface. What has been puzzling is how they propel themselves at such a high speed. They can’t pierce the water surface or they would sink. A team of mathematicians and engineers from the Massachusetts Institute of Technology (MIT)

applied conventional flow visualization techniques and high-speed video to examine in detail the movement of the water striders. They found that each stroke of the insect’s legs creates dimples on the surface with underwater swirling vortices sufficient to propel it forward. It is the rearward motion of the vortices that propels the water strider forward. To further substantiate their explanation, the MIT team built a working model of a water strider, called Robostrider, which creates surface ripples and underwater vortices as it moves across a water surface. Waterborne creatures, such as the water strider, provide an interesting world dominated by surface tension (see Problem 1.131).

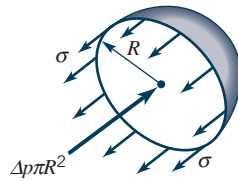


FIGURE 1.9 Forces acting on one-half of a liquid drop.

Table 1.5 (BG) and Table 1.6 (SI). Values for water as a function of temperature are provided in Appendix B (Tables B.1 and B.2). As indicated in the adjacent figure, the value of the surface tension decreases as the temperature increases.

The pressure inside a drop of fluid can be calculated using the free-body diagram in **Fig. 1.9**. If the spherical drop is cut in half (as shown), the force developed around the edge due to surface tension is $2\pi R\sigma$. This force must be balanced by the pressure difference, Δp , between the internal pressure, p_i , and the external pressure, p_e , acting over the circular area, πR^2 . Thus,

$$2\pi R\sigma = \Delta p \pi R^2$$

or

$$\Delta p = p_i - p_e = \frac{2\sigma}{R} \quad (1.21)$$

It is apparent from this result that the pressure inside the drop is greater than the pressure surrounding the drop. (Would the pressure on the inside of a bubble of water be the same as that on the inside of a drop of water of the same diameter and at the same temperature?)

Among common phenomena associated with surface tension is the rise (or fall) of a liquid in a capillary tube. If a small open tube is inserted into water, the water level in the tube will rise above the water level outside the tube, as is illustrated in **Fig. 1.10a**. In this situation we have a liquid–gas–solid interface. For the case illustrated there is an attraction (adhesion) between the wall of the tube and liquid molecules, which is strong enough to overcome the mutual attraction (cohesion) of the molecules and pull them up the wall. Hence, the liquid is said to *wet* the solid surface.

The height, h , is governed by the value of the surface tension, σ , the tube radius, R , the specific weight of the liquid, γ , and the *angle of contact*, θ , between the fluid and tube. From the free-body diagram of **Fig. 1.10b**, we see that the vertical force due to the surface tension is equal to $2\pi R\sigma \cos \theta$ and the weight is $\gamma\pi R^2 h$, and these two forces must balance for equilibrium. Thus,

$$\gamma\pi R^2 h = 2\pi R\sigma \cos \theta$$

so that the height is given by the relationship

$$h = \frac{2\sigma \cos \theta}{\gamma R} \quad (1.22)$$

The angle of contact is a function of both the liquid and the surface. For water in contact with clean glass $\theta \approx 0^\circ$. It is clear from Eq. 1.22 that the height is inversely proportional to the tube radius, and therefore, as indicated in the adjacent figure, the rise of a liquid in a tube as a result of capillary action becomes increasingly pronounced as the tube radius is decreased.

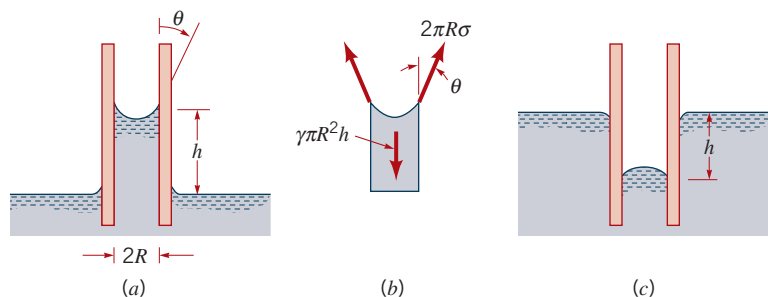
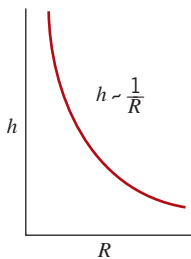


FIGURE 1.10 Effect of capillary action in small tubes. (a) Rise of column for a liquid that wets the tube. (b) Free-body diagram for calculating column height. (c) Depression of column for a nonwetting liquid.



Video

V1.10 Capillary rise



Video

V1.11 Contact angle

If adhesion of molecules to the solid surface is weak compared to the cohesion between molecules, the liquid will not wet the surface and the level in a tube placed in a nonwetting liquid will actually be depressed, as shown in **Fig. 1.10c**. Mercury is a good example of a nonwetting liquid when it is in contact with a glass tube. For nonwetting liquids, the angle of contact is greater than 90° , and for mercury in contact with clean glass $\theta \approx 130^\circ$.

Capillary action in small tubes, which involves a liquid–gas–solid interface, is caused by surface tension.

EXAMPLE 1.8 | Capillary Rise in a Tube

Given Pressures are sometimes determined by measuring the height of a column of liquid in a vertical tube.

Find What diameter of clean glass tubing is required so that the rise of water at 20°C in a tube due to capillary action (as opposed to pressure in the tube) is less than $h = 1.0\text{ mm}$?

Solution From Eq. 1.22

$$h = \frac{2\sigma \cos \theta}{\gamma R}$$

so that

$$R = \frac{2\sigma \cos \theta}{\gamma h}$$

For water at 20°C (from Table B.2), $\sigma = 0.0728\text{ N/m}$ and $\gamma = 9.789\text{ kN/m}^3$. Since $\theta \approx 0^\circ$, it follows that for $h = 1.0\text{ mm}$,

$$R = \frac{2(0.0728\text{ N/m})(1)}{(9.789 \times 10^3\text{ N/m}^3)(1.0\text{ mm})(10^{-3}\text{ m/mm})} \\ = 0.0149\text{ m}$$

and the minimum required tube diameter, D , is

$$D = 2R = 0.0298\text{ m} = 29.8\text{ mm} \quad (\text{Ans})$$

Comment By repeating the calculations for various values of the capillary rise, h , the results shown in **Fig. E1.8** are obtained. Note that as the allowable capillary rise is decreased, the diameter of the tube must be significantly increased. There is always some capillarity effect, but it can be minimized by using a large enough diameter tube.

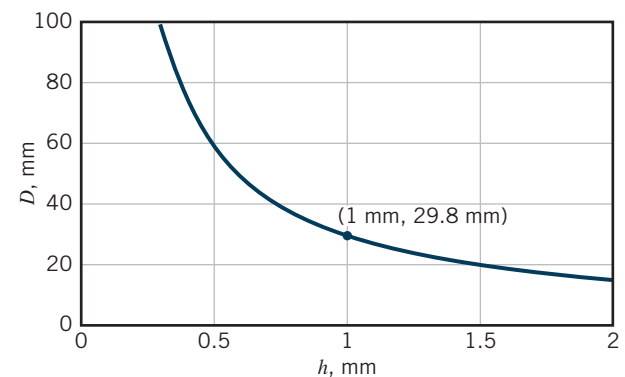
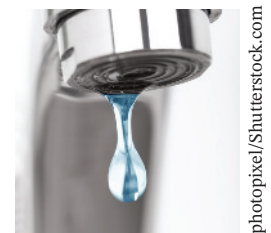


FIGURE E1.8

Surface tension effects play a role in many fluid mechanics problems, including the movement of liquids through soil and other porous media, flow of thin films, formation of drops and bubbles, and the breakup of liquid jets. For example, surface tension is a main factor in the formation of drops from a leaking faucet, as shown in the adjacent photograph. Surface phenomena associated with liquid–gas, liquid–liquid, and liquid–gas–solid interfaces are exceedingly complex, and a more detailed and rigorous discussion of them is beyond the scope of this text. Fortunately, in many fluid mechanics problems, surface phenomena, as characterized by surface tension, are not important because inertial, gravitational, and viscous forces are much more dominant.



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The Wide World of Fluids

Spreading of oil spills

With the large traffic in oil tankers there is great interest in the prevention of and response to oil spills. As evidenced by the famous *Exxon Valdez* oil spill in Prince William Sound in 1989, oil spills can create disastrous environmental problems. A more recent example of this type of catastrophe is the oil spill that occurred in the Gulf of Mexico in 2010. It is not surprising that much attention is given to the rate at which an oil spill spreads. When spilled, most oils tend to spread

horizontally into a smooth and slippery surface, called a slick. There are many factors that influence the ability of an oil slick to spread, including the size of the spill, wind speed and direction, and the physical properties of the oil. These properties include *surface tension*, *specific gravity*, and *viscosity*. The higher the surface tension the more likely a spill will remain in place. Since the specific gravity of oil is less than one, it floats on top of the water, but the specific gravity of an oil can increase if the lighter substances within the oil evaporate. The higher the viscosity of the oil, the greater the tendency to stay in one place.

1.10

A Brief Look Back in History

Some of the earliest writings that pertain to modern fluid mechanics can be traced back to the ancient Greek civilization and subsequent Roman Empire.

Before proceeding with our study of fluid mechanics, we should pause for a moment to consider the history of this important engineering science. As is true of all basic scientific and engineering disciplines, their actual beginnings are only faintly visible through the haze of early antiquity. But we know that interest in fluid behavior dates back to the ancient civilizations. Through necessity there was a practical concern about the manner in which spears and arrows could be propelled through the air, in the development of water supply and irrigation systems, and in the design of boats and ships. These developments were, of course, based on trial-and-error procedures. However, it was the accumulation of such empirical knowledge that formed the basis for further development during the emergence of the ancient Greek civilization and the subsequent rise of the Roman Empire. Some of the earliest writings that pertain to modern fluid mechanics are those of Archimedes (287–212 B.C.), a Greek mathematician and inventor who first expressed the principles of hydrostatics and flotation. Elaborate water supply systems were built by the Romans during the period from the fourth century B.C. through the early Christian period, and Sextus Julius Frontinus (A.D. 40–103), a Roman engineer, described these systems in detail. However, for the next 1000 years during the Middle Ages (also referred to as the Dark Ages), there appears to have been little added to further understanding of fluid behavior.

As shown in **Fig. 1.11**, beginning with the Renaissance period (about the fifteenth century), a rather continuous series of contributions began that forms the basis of what we consider to be the science of fluid mechanics. Leonardo da Vinci (1452–1519) described through sketches and writings many different types of flow phenomena. The work of Galileo Galilei (1564–1642) marked the beginning of experimental mechanics. Following the early Renaissance period and during the seventeenth and eighteenth centuries, numerous significant contributions were made. These include theoretical and mathematical advances associated with the famous names of Newton, Bernoulli, Euler, and d'Alembert. Experimental aspects of fluid mechanics were also advanced during this period, but unfortunately the two different approaches, theoretical and experimental, developed along separate paths. *Hydrodynamics* was the term associated with the theoretical or mathematical study of idealized, frictionless fluid behavior, with the term *hydraulics* being used to describe the applied or experimental aspects of real fluid behavior, particularly the behavior of water. Further contributions and refinements were made to both theoretical hydrodynamics and experimental hydraulics during the nineteenth century, with the general differential equations describing fluid motions that are used in modern fluid mechanics being developed in this period. Experimental hydraulics became more of a science, and many of the results of experiments performed during the nineteenth century are still used today.

At the beginning of the twentieth century, both the fields of theoretical hydrodynamics and experimental hydraulics were highly developed, and attempts were being made to unify the two. In 1904, a classic paper was presented by a German professor, Ludwig Prandtl (1875–1953), who introduced the concept of a “fluid boundary layer,” which laid the foundation for the unification

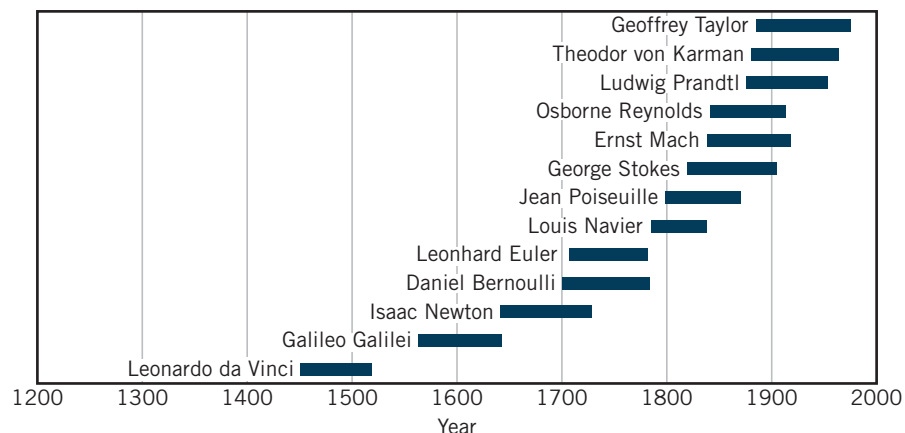


FIGURE 1.11 Time line of some contributors to the science of fluid mechanics.

of the theoretical and experimental aspects of fluid mechanics. Prandtl's idea was that for flow next to a solid boundary a thin fluid layer (boundary layer) develops in which friction (viscous force) is important, but outside this layer the fluid often behaves very much like a frictionless fluid. This relatively simple concept reconciled the seemingly different conclusions between the hydrodynamicists and the hydraulicists. Prandtl is generally accepted as the founder of modern fluid mechanics.

Also, during the first decade of the twentieth century, powered flight was first successfully demonstrated with the subsequent vastly increased interest in *aerodynamics*. Because the design of aircraft required an understanding of fluid flow and an ability to make accurate predictions of the effect of airflow on bodies, the field of aerodynamics provided a great stimulus for the many rapid developments in fluid mechanics that took place during the twentieth century.

As we proceed with our study of the fundamentals of fluid mechanics, we will continue to note the contributions of many of the pioneers in the field. **Table 1.9** provides a chronological

The rich history of fluid mechanics is fascinating, and many of the contributions of the pioneers in the field are noted in the succeeding chapters.

TABLE 1.9 Chronological Listing of Some Contributors to the Science of Fluid Mechanics Noted in the Text

ARCHIMEDES (287–212 B.C.)

Established elementary principles of buoyancy and flotation.

SEXTUS JULIUS FRONTINUS (A.D. 40–103)

Wrote treatise on Roman methods of water distribution.

LEONARDO da VINCI (1452–1519)

Expressed elementary principle of continuity; observed and sketched many basic flow phenomena; suggested designs for hydraulic machinery.

GALILEO GALILEI (1564–1642)

Indirectly stimulated experimental hydraulics; revised Aristotelian concept of vacuum.

EVANGELISTA TORRICELLI (1608–1647)

Related barometric height to weight of atmosphere, and form of liquid jet to trajectory of free fall.

BLAISE PASCAL (1623–1662)

Finally clarified principles of barometer, hydraulic press, and pressure transmissibility.

ISAAC NEWTON (1642–1727)

Explored various aspects of fluid resistance— inertial, viscous, and wave; discovered jet contraction.

HENRI de PITOT (1695–1771)

Constructed double-tube device to indicate water velocity through differential head.

DANIEL BERNOULLI (1700–1782)

Experimented and wrote on many phases of fluid motion, coining name “hydrodynamics”; devised manometry technique and adapted primitive energy principle to explain velocity-head indication; proposed jet propulsion.

LEONHARD EULER (1707–1783)

First explained role of pressure in fluid flow; formulated basic equations of motion and contributed to the so-called Bernoulli theorem; introduced concept of cavitation and principle of centrifugal machinery.

JEAN le ROND d'ALEMBERT (1717–1783)

Originated notion of velocity and acceleration components, differential expression of continuity, and paradox of zero resistance to steady nonuniform motion.

ANTOINE CHEZY (1718–1798)

Formulated similarity parameter for predicting flow characteristics of one channel from measurements on another.

GIOVANNI BATTISTA VENTURI (1746–1822)

Performed tests on various forms of mouthpieces—in particular, conical contractions and expansions.

LOUIS MARIE HENRI NAVIER (1785–1836)

Extended equations of motion to include “molecular” forces.

AUGUSTIN LOUIS de CAUCHY (1789–1857)

Contributed to the general field of theoretical hydrodynamics and to the study of wave motion.

GOTTHILF HEINRICH LUDWIG HAGEN

(1797–1884)

Conducted original studies of resistance in and transition between laminar and turbulent flow.

JEAN LOUIS POISEUILLE (1799–1869)

Performed meticulous tests on resistance of flow through capillary tubes.

HENRI PHILIBERT GASPARD DARCY

(1803–1858)

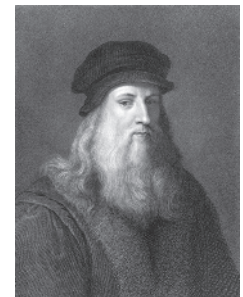
Performed extensive tests on filtration and pipe resistance; initiated open-channel studies carried out by Bazin.

JULIUS WEISBACH (1806–1871)

Incorporated hydraulics in treatise on engineering mechanics, based on original experiments; noteworthy for flow patterns, nondimensional coefficients, weir, and resistance equations.

WILLIAM FROUDE (1810–1879)

Developed many towing-tank techniques, in particular the conversion of wave and boundary layer resistance from model to prototype scale.



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Ludwig Prandtl

TABLE 1.9 Chronological Listing of Some Contributors to the Science of Fluid Mechanics Noted in the Text (continued)

ROBERT MANNING (1816–1897) Proposed several formulas for open-channel resistance.	EDGAR BUCKINGHAM (1867–1940) Stimulated interest in the United States in the use of dimensional analysis.
GEORGE GABRIEL STOKES (1819–1903) Derived analytically various flow relationships ranging from wave mechanics to viscous resistance—particularly that for the settling of spheres.	MORITZ WEBER (1871–1951) Emphasized the use of the principles of similitude in fluid flow studies and formulated a capillarity similarity parameter.
ERNST MACH (1838–1916) One of the pioneers in the field of supersonic aerodynamics.	LUDWIG PRANDTL (1875–1953) Introduced concept of the boundary layer and is generally considered to be the father of present-day fluid mechanics.
OSBORNE REYNOLDS (1842–1912) Described original experiments in many fields—cavitation, river model similarity, pipe resistance—and devised two parameters for viscous flow; adapted equations of motion of a viscous fluid to mean conditions of turbulent flow.	LEWIS FERRY MOODY (1880–1953) Provided many innovations in the field of hydraulic machinery. Proposed a method of correlating pipe resistance data that is widely used.
JOHN WILLIAM STRUTT, LORD RAYLEIGH (1842–1919) Investigated hydrodynamics of bubble collapse, wave motion, jet instability, laminar flow analogies, and dynamic similarity.	THEODOR VON KÁRMÁN (1881–1963) One of the recognized leaders of twentieth century fluid mechanics. Provided major contributions to our understanding of surface resistance, turbulence, and wake phenomena.
VINCENZ STROUHAL (1850–1922) Investigated the phenomenon of “singing wires.”	PAUL RICHARD HEINRICH BLASIUS (1883–1970) One of Prandtl’s students who provided an analytical solution to the boundary layer equations. Also demonstrated that pipe resistance was related to the Reynolds number.

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listing of some of these contributors and reveals the long journey that makes up the history of fluid mechanics. This list is certainly not comprehensive with regard to all past contributors but includes those who are mentioned in this text. As mention is made in succeeding chapters of the various individuals listed in Table 1.9, a quick glance at this table will reveal where they fit into the historical chain.

It is, of course, impossible to summarize the rich history of fluid mechanics in a few paragraphs. Only a brief glimpse is provided, and we hope it will stir your interest. References 2 to 5 are good starting points for further study, and in particular, Ref. 2 provides an excellent, broad, easily read history. Try it—you might even enjoy it!

Chapter Summary and Study Guide

Chapter Summary

This introductory chapter discussed several fundamental aspects of fluid mechanics. Methods for describing fluid characteristics both quantitatively and qualitatively are considered. For a quantitative description, units are required. The concept of dimensions is introduced in which basic dimensions such as length, L , time, T , and mass, M , are used to provide a description of various

quantities of interest. The use of dimensions is helpful in checking the generality of equations, as well as serving as the basis for the powerful tool of dimensional analysis discussed in detail in Chapter 7.

Various important fluid properties are defined, including fluid density, specific weight, specific gravity, viscosity, bulk modulus,

speed of sound, vapor pressure, and surface tension. The ideal gas law is introduced to relate pressure, temperature, and density in common gases, along with a brief discussion of the compression and expansion of gases. The distinction between absolute and gage pressure is introduced. This important idea is explored more fully in Chapter 2.

The following checklist provides a study guide for this chapter. When your study of the entire chapter and end-of-chapter exercises has been completed, you should be able to

- write out meanings of the terms listed and understand each of the related concepts. These terms are particularly important and are set in **bold black** type in the text.
- determine the dimensions of common physical quantities.
- determine whether an equation is a general or restricted homogeneous equation.
- correctly use units and systems of units in your analyses and calculations.
- calculate the density, specific weight, or specific gravity of a fluid from a knowledge of any two of the three.
- calculate the density, pressure, or temperature of an ideal gas (with a given gas constant) from a knowledge of any two of the three.
- relate the pressure and density of a gas as it is compressed or expanded using Eqs. 1.14 and 1.15.
- use the concept of viscosity to calculate the shearing stress in simple fluid flows.
- calculate the speed of sound in fluids using Eq. 1.19 for liquids and Eq. 1.20 for gases.
- determine whether boiling or cavitation will occur in a liquid using the concept of vapor pressure.
- use the concept of surface tension to solve simple problems involving liquid–gas or liquid–solid–gas interfaces.

Key Terms

fluid
units
basic dimensions
dimensionally homogeneous
density
specific weight
specific gravity

ideal gas law
absolute pressure
gage pressure
no-slip condition
rate of shearing strain
absolute viscosity
Newtonian fluid

non-Newtonian fluid
kinematic viscosity
bulk modulus
speed of sound
vapor pressure
surface tension

Key Equations

Specific weight	$\gamma = \rho g$	(1.6)
Specific gravity	$SG = \frac{\rho}{\rho_{\text{H}_2\text{O}@4^\circ\text{C}}}$	(1.7)
Ideal gas law	$p = \rho RT$	(1.8)
Newtonian fluid shear stress	$\tau = \mu \frac{du}{dy}$	(1.9)
Bulk modulus	$E_v = -\frac{dp}{d\psi/\psi}$	(1.12)
Speed of sound in an ideal gas	$c = \sqrt{kRT}$	(1.20)
Capillary rise in a tube	$h = \frac{2\sigma \cos \theta}{\gamma R}$	(1.22)

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